

AVIATION WEEK

Oct. 23, 1950

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A YEAR

A MCGRAW-HILL PUBLICATION



Francis Fox at Worcester KNOWS!

Worcester, Mass., was one of the first city-owned airports with L-M High Intensity Lighting on all runways. Manager Francis T. Fox and members of the City Airport Commission made a very careful check of lighting and the experience of other airport managers before the decision was made. Captain Fox, a pilot himself, recently told us: "We've had nearly three years of operation now with high intensity lights on all runways. Our experience has convinced us that our choice was a wise one, and our lights have proved their value with better operations in all kinds of weather."



Ask the men who KNOW L-M high intensity runway lighting

Ask airport managers, airline men, and pilots who use and know L-M high intensity runway lighting. They can tell you from their own experiences and observation how important it is to have the 180,000 beam candle power, the freedom from halo and glare, that only L-M lighting offers, with its extremely high intensity and controllable beam. Then ask the L-M Field Engineer for details or write Airport Lighting Division, Line Material Co., East Stroudsburg, Pennsylvania (a McGraw Electric Company Division).



L-M's 180,000 cp. high intensity runway light with the famous controllable beam.



J. E. Hightower at Knoxville KNOWS!

"Here in the Tennessee mountains, we get all kinds of weather," says J. E. Hightower, manager of Knoxville's McGhee Tyson Airport. "And not all of it is good flying weather. So we are very much concerned with the high penetration of our lights, so that we can give the pilot the best possible delineation of the runway. Our experience with the lights has been most gratifying, and an important factor in maintaining a good safety record here."

Vic Dallin at Philadelphia KNOWS!

"Whenever other eastern airports are closed by weather and our Philadelphia International Airport is marginal, all pilots appreciate the great advantage of the controllable-beam high intensity runway lights," says J. Victor Dallin, chief of Philadelphia's Bureau of Aeronautics. "We have had as many as 79 airliners in a single day take refuge here due to weather conditions. We are presently extending our instrument runway and these lights will naturally extend the runway and naturally this extension will be made with these lights."

LINE MATERIAL... Airport



YOU CAN BE **SURE**... IF IT'S
Westinghouse



All twin engine fighters for the Navy's newest carrier-based squadrons are powered by the J-34. This light and sleek Westinghouse engine leads itself ideally to a twin engine installation which in turn provides the restoring safety factor of single engine operation in times of emergency.

The designers of these airplanes chose the J-34 because it combines high power with low weight. Three features plus the power, dependability and performance of the engine make it the striking force of the United States Navy will be second to none.

F 24000 2

Westinghouse
AVIATION
GAS TURBINES



B.F. Goodrich



How to get more landings out of a lining

A new kind of brake block that lasts longer, provides better braking and saves weight has been developed by B. F. Goodrich. It is now in use on the C-124 Globemaster II (above) built by the Long Beach Plant of Douglas Aircraft Company, and on the B-36, B-45 and B-47.

Secret of the new brake block is revealed in the three photos above. There are no rivets. Instead the brake lining is cemented onto a special magnesium disc with a new, super-strong B. F. Goodrich armor.

Manufacture of the discs makes it possible to use more of the brake lining. You get full, positive braking

down almost to the metal backing!

The magnesium backing also makes the brake block more rigid, providing full, even contact between lining and drum for better braking, slower wear. The magnesium disc is perforated for more rapid dissipation of heat. And this construction is both lighter and stronger than the rivet type.

The new design B. F. Goodrich brake also has a narrow cavity expansion tube that gives greater braking pressure with less heat. And a new spider-type frame that provides extra strength with less weight.

The basic BFG expansion tube design offers still other advantages. Quicker,

easier maintenance. No locking or gobbling. Less weight for a given amount of kinetic energy than any other brake design. Ability to take emergency overloads better. Longer life.

Guaranteed improvement of B. F. Goodrich brakes is a typical example of how BFG research works for you. It pays to specify "B. F. Goodrich." The B. F. Goodrich Company, American Division, Akron, Ohio.

B.F. Goodrich
FIRST IN RUBBER

FLIGHT ECONOMY



Western Air Lines assures it by using TEXACO

High dependability and economy are "musts" for profitable operation . . . and Texaco assures both for Western Air Lines.

Texaco Aerojet Engine Oil, for example, assures the clean operation that means dependability of performance with lower fuel consumption and lower maintenance costs. It's the industry's first choice for engine lubrication. In fact—

More revenue airline miles in the U. S. are flown with Texaco Aerojet Engine Oil than with any other brand!

In addition, Texaco's advanced Lubrication Engi-

n in 1958, Western Air Lines made the first scheduled commercial airline flights with regular runs between Los Angeles and Salt Lake City, New, stock Western Air Lines Economy DC-4 Conquerors and Western's De Luxe Constellation took every major city on the Pacific Coast with a single schedule that makes commuting easier. All Western Air Lines planes serving throughout its 44-city system in the West are lubricated with Texaco Aerojet Engine Oil exclusively—and have been for years.

neering Service is ready with many suggestions and proved ideas to simplify service and lubrication procedures that further keep maintenance costs at a minimum.

Contact a Texaco Aviation Representative for invaluable advice and assistance. Just off the coast of the state are then 2,000 Texaco Wholesale Distributing Places in the 48 States, or write The Texaco Company, Aviation Division, 135 East 42nd Street, New York 17, N. Y.



TEXACO Lubricants and Fuels

FOR THE AVIATION INDUSTRY

TUNE IN: TEXACO STAR THEATRE playing MILTON BIBLE on television every Tuesday night. See newspaper for time and station.

NEWS DIGEST

DOMESTIC

An Traffic Conference of the Air Transport Assn. elected the following airline representatives as its officers for 1959: president, Walter Striebing, vice president—traffic and sales for National; first vice president, James W. Austin, vice president—traffic and sales; Capital, second vice president, Harding Lamm; vice president—traffic, Pioneer.

Air Force appointments made by the President included that of Lt. Gen. Nefert Y. Young as Vice Chief of Staff with the rank of four star general. Lt. Gen. Loren Noyd was appointed Commander-in-Chief of the U. S. Air Force in Europe, succeeding Lt. Gen. John K. Cannon. Lt. Gen. John H. Johnson was appointed Deputy Chief of Staff, Operations. He had been acting DCSO since May 1958. Gen. Norford had been Acting Vice Chief of Staff. Gen. Cannon, whom he succeeds, has been named Commanding General of the Tactical Air Command at Langley AFB, Va.

An employment agreement covering about 3200 Pan American Airways pilots has been signed by the airline with the Air Line Pilots Assn. The contract is to be effective from Nov. 1, 1958 through Dec. 31, 1971, providing for increases for pilots and copilots, vacation, travel for themselves and improved working conditions. Pilots will receive a guaranteed year plus extra money depending on aircraft weight and speed, with pay increases after 70 hours a month instead of the previous 80 hours. Copilot pay will range from \$300 a month to \$700 a month for 70 hours work, compared with \$275 to \$625 for 56 hours under the previous contract.

Joseph J. O'Connell, Jr., former CAB chairman, has joined a law firm—now Chapman, Byrnes, Walsh & O'Connell—with offices in New York and Washington. The Washington office opened Oct. 1.

A Northwest Airlines 3-0-2 crashed near Alameda, Mo., killing five of the six men aboard for a motor crash. Right: The plane took off from Minneapolis, 50 miles away.

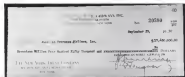
Major Gen. Cyril A. Anderson, former head of the Air War College at Montgomery, Ala., has been selected to the command of the 778th Tactical Training Wing, Shreveport AFB, Wichita Falls, Tex. Gen. Anderson was transferred from the War College just for allegedly leaving "positive war."

The American Airlines DC-6 that landed twice last August after a propeller and engine lost loose and ejected a hole in its fuselage at it flew over the Rockies is being test flown after preliminary repairs.

Striking engineers at Wright Aeronautical Corp., Wood Ridge, N. J., reported earlier plans to return to work accompanied by company offers of salary increases. Neither of two workers' groups involved would accept limitations of grievance procedure and a request that contract negotiations be by mail in December and go over to December 1959.

FINANCIAL

Aeromop Corp. has declared a quarterly dividend of 5 cents per common stock share, payable Nov. 15 to stockholders as of Nov. 1.



BIGGEST CHECK in the history of airline transactions was paid by Pan American World Airways to American Overseas Air Lines last month when PAA took over

AOA's routes, assets and routes. Pictured above, the \$27,400,000 document was signed by two PAA officials—Comptroller J. S. Wadsworth and Treasurer R. G. Ferguson.



CP Hot Dimpler 520 EA

for hot dimpling

of magnesium and the harder aluminum alloys

In dimpling magnesium and the harder aluminum alloys, the application of heat is recommended to eliminate cracked dimples.

Developed for this type of work, the CP Hot Dimpler incorporates Zephyr cone edge dimpling punches and dies which assure accurate forming of dimples. Write for detailed information on the new CP-520-EA hot dimpler.



AIR COMPRESSORS • AIRCRAFT AIRLIFTERS • PNEUMATIC AND HYDRAULIC WELDERS • HYDRAULIC DIVIDERS • ELECTRIC DRILLS

INDUSTRY OBSERVER

►Fairechild is expected to license its C-319 Pallet transport for military construction in England and France, with sales restricted to Sterling Rice countries.

►In spite of the huge orders which Republic Aviation has for its F-54 jet fighter, approximately twice as many more may be built overseas by French and British plants under license as part of the Mutual Defense Assistance Program.

►While Curtiss-Wright's Propeller division has possessed the technology of using reverse propeller pitch for extremely rapid emergency descent, both Hamilton Standard and Aeromodular are now getting into the act, as the technology gets greater acceptance by the military. First military airplanes to use it are expected to be the Boeing B-55 and Convair B-56 bombers, and the Navy's Grumman F9F and Chance Vought F4U fighters.

►An aircraft equipment order is pending that CAA's dipping of all 7 precision approach radars from the 1951 federal aviation program is a revival of the old CAA position on radar, which led to the violent ILS CAA feud in government agencies.

►First two Army Canada CF-105 two-seat night fighters have now taken up over 50 flights, while two of Army Canada's SF06H three-seat turboprop engines have made more than 20 flights in their Lancaster flying test bed installation. Canada's annual test runs now total more than 3000 h—one engine has already had about 750 h of test run without scratch.

►Continuing of the four-place Dutch Panavia air two plane, developed by Fokker and now built by Royal Dutch aircraft factories, is expected to be lifted shortly following thorough investigation into the matter of propeller vibration troubles which have resulted in the breaking of the shafts of the air two craft.

►Observers who looked at the British Sapphire jet engine, on display at the Pentagon recently along with the Armstrong Siddeley Mamba and Python turboprops, say that the Sapphire displayed was not complete enough to give away much engineering details.

►At least two American engine companies are looking into ducted fan developments in the next step beyond propeller. Idea is to use a multi-stage duct-bladed fan something like a jet engine compressor, raised by a unit comparable to a turboprop engine, and all thrusting into one duct-like casing which made a turbine.

►Consolidated Vultee now has about one-fourth of its B-36 program subcontracted to other manufacturers in such an extensive program that a new subcontract department has been set up to handle it at St. Louis. Principal subcontractors, in addition to Bell Aircraft which has been making jet engine tail nozzles for the B-36D since this model was started, are: Beech, making lower nacelles, landing gear doors, radomes, vertical fin; Sperry (Tulsa), making engine exhaust extensions, turret doors and dividers; Inland Industrial (Oakland, Tex.), making wing center section trailing edges; and Teledyne (Dallas), making horizontal stabilizers.

►National Airlines is "planning" up the DC-66s it ordered in early 1960 from Douglas for fast-lift daytime service, selecting it has accepted, however unwillingly. CAA's decision against the Douglas DC-66 New York-Miami route may National had twice delivery on all but one of the five planes, and it was too far along to change the order at the Douglas plant. But National is prepared to reconnect them to 68 passenger coaches anywhere that the CAA gives the authorization for the day-light coach service.

WHO'S WHERE

In the Front Office

Since South has been made director of engineering of Bess Aircraft Co. South stepped up from the position of chief engineer of the firm's airplane division. Before coming to America in 1948, he ran chief design engineers for 5 years at General Electric engine under South will be W. T. H. Meadows, formerly project engineer.

Malcolm G. Montgomery has been appointed director of sales for Pong, Dart Lee, according to Albert J. Jensen, who is now in duties as assistant to the president. Montgomery has had 15 years of design and design experience with trucks and aircraft and served in transportation after the Fourth Air Force during the war. Jensen was a KLM and Sabena & Western since the.

President of A. H. Engstrom is acting engineering manager for American Helicopter Plant. Since made by the Metal Products division, has been removed by the Koppert Co. He was assistant to the manager of the designing and research department.

Changes

Col. J. A. Villanor, consultant on Air Test system to the Administrator of Civil Aeronautics, has stepped down from the CAA to go back on active duty with the Air Force. Villanor formerly was Administrator of the CAA of the Philippines and general manager of the National Aeronautics Corp. GAO also announced the resignation of GAO A. Gilbert, as traffic control expert, to enter private business.

John L. Stupp Ziegler has been named chief of flight test at Bell Aircraft. A rocket engine test pilot at Bell Aircraft, Ziegler returns from South American Aviation where he was in a year as a pilot engineer on the F-96, B-47 and A-1. He formerly was a production test pilot for Curtiss Wright.

What They're Doing



RESTING ON HIS LAURELS J. W. Lott, Convair's chief engineer at Ft. Worth, matches cut on one of the two new books being added in the forward cabin of B-36D bombers. Without those, crews would have to fight their seat time between on books.

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IMMEDIATE PRODUCTION for a limited quantity will be started on XC-122C, one winner in the assault transport game, and . . .



FUTURE PRODUCTION in quantity will be started on Curtiss-Wright XC-123, both planes to be produced in Birmingham.

Assault Transport Order Goes to Chase

With both its entries in the competition winners, company moving to larger plant.

By Ben S. Lee

Chase Aircraft Co. is the preselected winner of the assault transport competition and will move from Yonkers, N. Y., to produce the XC-123C and XC-123D assault transports at the former Bechtel-McGraw-Hill B-24 modification plant at Birmingham, Ala., according to sources here.

As a result of its success in the competition which pitted it against Northrop Aircraft (C-123A Barden) and Fairchild Engine and Airplane Co. (stripped down C-123D), Chase Air-

craft will be asked for immediate but limited production of the XC-123C, while it tools up for quantity production of the XC-123D.

►Assault Competition—The assault transport, principal point of contention between Army and Air Force over roles, needs and technology in the tactical support of ground troops was a hot issue during "Exercise Overcast" conducted in the Caribbean this summer. Success of the Phase I problems in the combined maneuver of Swanton—initial assault assault—was doubtful, according to critics. Maintaining the actual

(AVIATION WEEK MAY 6) also presented many unprecedented battle problems.

Lessons learned at Swanton, strengthening primarily tactical Army plans, indicated that an assault transport was urgently required in support of the rapid Army Air Force. Recent lessons learned the hard way in Korea (Aviation Week Aug. 23) have strengthened that conviction.

As a result of Swanton, the long delayed assault transport evaluation competition was ordered and took place in rapid sequence at Wright-Patterson AFB, Ohio, Elgin AFB, Ill., and at Ft. Rucker, N. C.

►Thorough Evaluation—The actual evaluation of the Northrop C-123A, the modified Fairchild C-123, Chase's XC-123C, XC-123D and C-123 and C-123

122C, XC-121 and G-48 and G-20 gliders took place at Eglin and proved a milestone in test technique (Aeronautics Week Sept. 11).

All entries reflected considerable design changes made and tested, and all problems. Eglin test flight personnel stand there in consternation with military engineers the test included maneuvers probably more rigorous than an actual transport would likely face in actual battle.

The tactical evaluation, conducted at Eglin and later at Ft. Rucker, sought to prove among other special considerations that a powered aircraft could get into and out of the test area and, as well as could a low-altitude glider under combat conditions.

Army has long been concerned with the high loss rates of personnel and equipment in the transport category. "This is borne out by statistics of loss in combat operations (captured) by gliders and infantry teams during World War II."

Seriously, Army is concerned about high loss rates of transport aircraft, and is looking for ways to make them more resistant to aerial operations. It admits high priority for aerial defense against aerial troops, but deplores that so far it must depend entirely upon parachuted troops to secure an area.

► **Conversion Scrapped** — Originally, USAF had early considerations as to whether to convert a number of the obsolescent C-47 Packard turbo engines to smooth transport category. The plan is now said to be shifted in favor of immediate production of the XC-122C for a number of reasons. Among them is the urgent need of C-47s now available for potentially assigned missions of troop and cargo transport. Another factor opposing conversion of a "useable quantity" of the C-47s is cost. It has been estimated that it would require from \$15,000 to \$40,000 each to modify transport modification. USAF sources consider this cost excessive for a temporary expedient.

► **Northrop**—Sources close to top Northrop Aircraft management report that the C-115A has already been item of company talks in a bid. Northrop had hoped to convince USAF that the C-115, developed originally as a lighter version to capture the short haul, can meet transport needs, could be converted to meet Army Field Forces need for assault transport. USAF now has a service quantity of C-115s in Arctic rescue operation.

While award of the assault transport contract to Northrop would have come as an "ideal" plan, contracts for nearly 900 F99 Scorpion jet fighters out of Ford '50 and '51 regular and supply aircraft budgets, not counting acquisition and production for 1952 and 1953, will leave Northrop at previously planned production levels.

► **Fairchild-McDonnell**, officials of Fairchild, while watching the entry of another manufacturer in the medium transport field, are hard at work at the design of a new four-engine transport to replace replacement for the C-119. Air Force is now capitalizing equipment for lack of a transport.

C-119 production schedule for both domestic military requirements and to support troop and cargo transport needs of missions allied under the North Atlantic Treaty are exceedingly heavy. Plans are fully lined up for acquisition of another assembly plant to speed production. If Fairchild is awarded some of USAF's longer production contract, a second plant will be essential to meet production schedules of both C-119 and C-71.

► **Chase-Whitcomb** of the Birmingham, Ala., 24-a cabinizer plant to Chase Aircraft has not yet been fully accepted. That Chase will certainly need larger quarters for planned USAF production of the C-122 and C-123. Company officials state early that it is negotiating for larger facilities to handle planned expanded production.

Washington sources indicate that Chase is very probably scheduled for Birmingham.

XC-122C, the lighter of the two Chase assault transports is a twin-engine plane version of the DC-8 glider. The plane features air ramp loading. The ramp operates by hydraulically actuated

tail door doors. The main door opens downward and forms a shallow ramp for vehicle loading. Door may be opened in flight for pre-dumping and can be rapidly closed.

Heavy shipments of vehicles without power can be pulled aboard by means of a cable running through the craft to a power source. Provisions for loading of truck bed height are also incorporated. Reversible jettison of the XC-122C enables the plane to back into loading area to receive or discharge typical assault loads. These include: A one and one-half ton truck, 18-wheel, 10-wheel, plus one jeep or comparable combination of selected loads and by the second floor.

For evacuation of wounded the plane can transport 24 litter patients and two medical attendants and all their equipment.

Capable of being towed by other aircraft from airfields or by switch package, the plane incorporates both tow and tow equipment in one and the same. Span of the XC-122C is 95 ft. 1 in.; length, 82 ft. 8 in.; height to tip top, 24 ft. 3 in. Cargo compartment dimensions are height, 6 ft. 6 in.; length, 71 ft. 8 in.; width, 7 ft. 8 in. Usable floor area, 240 sq. ft.; usable ceiling, 1500 ft.

Weight empty, 19,000 lb.; max gross, 42,000 lb.; max useful load, 21,000 lb. Top speed of the XC-122C is 240 mph; cruising speed, 200 mph; landing speed, 85 mph; range, 1070 mi. with 670 gal. fuel, service ceiling, 18,100 ft.

Forwardmost consists of two Wright R-1820-101 engines developing 1425 hp, at 2700 rpm at takeoff.

Provisions for fuel tank, fuel handling, venting, three bleed, of 12 ft. 6 in. dia.

XC-123, heavier Chase assault transport for which USAF will ask more production than its lighter counterpart by a nearly 5000 lb. Developed specifically to fit Army's requirements for an assault transport as well as to meet Army USAF transport requirements, the XC-123 incorporates steel-welded steel nose structure, overhead a heavy-duty diving aircraft emergency, cockpit floor must have the cargo compartment floor with a heavy built-in bulkhead between cockpit and cargo compartment in case of a cargo shift and personnel's self-protecting seat belts.

The plane has a rear loading ramp for cargo loading. The ramp can be lowered or raised by intermediate position for direct loading of cargo from truck trailer bed or cargo platform.

Construction from steel cannot be towed by other aircraft but would not be acceptable without reinforcement with self-contained equipment.

As a personnel carrier the XC-123 can carry 30 fully equipped troops or 50 litter patients, 6 ambulatory patients and 6 medical attendants.

Designed to carry a useful load of 27,200 lb., the plane includes stretch bed of a low release mechanism in both the nose and tail, as it is designed to operate as a low plane, as in a towed aircraft, either with or without engine action.

Span of the XC-123 is 110 ft.; length, 77 ft. 2 in.; height to tip top, 32 ft. 8 in. Cargo compartment dimensions are height, 8 ft. 2 in.; length, 61 ft. 8 in.; width, 9 ft. 2 in.; usable floor area, 456 sq. ft.; usable ceiling of the XC-123 is 1570 ft.

Weight empty is 26,000 lb.; normal gross wt., 54,000 lb.; useful load, 27,100 lb.; max gross wt., 78,000 lb.; useful load, 44,000 lb.

It will meet construction, the XC-123 is powered by two Pratt and Whitney

R-1800 C-14 engines rated at 2100 hp each on takeoff at 2600 rpm. Propellers are constant-speed, full-authority, reversible, three-bladed, 15-ft. Hamilton Standard.

Top speed of the XC-123 is 245 mph; cruising speed, 205 mph; landing speed, 85 mph; range, 1160 mi.; service ceiling, 28,000 ft.

Bristol Proteus Will Be Built in U.S.

An authoritative Defense Department spokesman has told *Aeronautics Week* that a second powered British gas turbine, manufactured by Bristol, has been approved for manufacture by Curtiss-Wright Co. The engine, which is to be used in the Proteus, single and twin power pack, figures in the British package engine deal. It is reported that a new Bristol engine design is the pre-

vious situation in the assignment. The Proteus single-engine turbine engine has a sea level power of 1200 shaft hp, gear 800 hp, static jet thrust. The twin Proteus develops 1800 shaft hp, static jet thrust. The engine Proteus engines are scheduled to provide power packs for two of Great Britain's largest assault—the Bristol Hercules Mk.2 and the Saunders Roe Princess flying boat.

South Africa Tours

(McGraw-Hill World News)

Johannesburg—South African Airways will offer the public an excursion to enable them to undertake a 15-day aerial trip in Africa, the United States and Canada to attend next shipping place will be London, New York, and Montreal. The excursion is a result of the recent air conference in Madrid.



AT FIRST HAND

First-hand information on the operation of Boeing jet for Southern Transport Co. is being given to United Statesmen John A. McGraw jet first-hand information on the operation of a Boeing B-47 Stratojet recently during a flight over Chesapeake Bay at Ft. Worth, Tex. The Boeing handled the jet's attack during part of the flight. McGraw was on the wingman's post. Fisher shows the jet in flight over Chesapeake Bay.



SUPER CONSTELLATION shows its greater length when parked on the Lockheed ramp beside a conventional Model 749.

Lockheed Shows First Super-Constellation

Lockheed Aircraft Corp. apparently has decided its future course in transport aircraft. Now the Super-Constellation, first with compound engines, then with turbo-prop power, and a pure jet transport.

There will be no Lockheed transport developed especially for the transport, at least according to present thinking. And turbo-prop power on the Super-Constellation may be merely an interim step until engines were ready for pure jet.

Lockheed President Allan G. Smith told *Aeronautics Week* that he brought an aircraft development should have a guaranteed order of 25 to 30 planes before undertaking to produce a jet transport. But he added that Lockheed might go ahead if it had an order for 25 to 30 such planes.

The Lockheed is currently weighing a jet transport project now evident last week at the company made its first official announcement of its Super-Constellation in connection with the meeting in San Francisco of the International Air Transport Assn. (See page 11.) It called the Super-Constellation (Model 149C) as "designed to handle the job between medium planes and the first American jet transport."

The Super-Constellation, 18 ft. longer than the Model 749, was due to be flown for NATS deliveries when they moved Los Angeles. In order to speed the plane's development, Lockheed modified its original Model 049. The first production plane is scheduled to fly next April. To date, the company has \$50 million worth of orders from the Air Force,

Navy, Eastern Air Lines and KLM. First of the Super-Constellation will be powered by Wright R-3350 C-18CA-1 engines which, with water injection, will develop more than 2100 hp each. Later versions will use the Wright R-3350-DA compound engines which develop up to 1800 hp each. The compound engine, says Lockheed, will permit a gross weight of 132,000 lb. and 40 percent more payload than the Model 749.

With a fuselage 111 ft. 7 in. long, the Super-Constellation will carry 75 first-class passengers, or accommodate up to 118 coach passengers. Use of compound engines will increase range, depending on speed and load, and in addition the Model 149C is designed to carry sleeping fuel tanks for extra emergency range.



SKATE 1 is one of the three major model changes that have taken place in Convair's years of research on flying boat designs.

Design for a Supersonic Flying Boat?

Years of research has put Convair in position to go ahead with prototype as soon as Navy gives signal.

By Alexander McManis

Almost everywhere in aviation these days, supersonic is inherently slower than subsonic. But a small group of American designers, who question the inflexibility of that premise, has been working for years to build a case to disprove. Having a delay due to immediate plane requirements by Navy—these question a prototype high-speed airplane embodying their design may be long within two years.

► **Supersonic Supersonic**—The Navy calls their highly unorthodox undertaking Project Skate. When it is launched and flown as a full scale prototype developed by Convair's Vulture, it may well be the world's first supersonic flying boat. Already the Skate has passed through at least nine major model changes and refinements to reach the stage now approved for full scale prototype construction.

► **Shaping Progress**—But that is only a fraction of the experimenting and testing which has gone into the long painstaking program to show down the bulky old subsonic airplane, into a streamlined supersonic form.

It took to design a supersonic plane to begin with. But when you have to make it seaworthy to boot, there are really multiply more problems.

► **Convair's Design**—High point of the case, supersonic flying boat design isn't cut in a single-day, hasty. Convair engineers, Ernest G. Stout, Junior, on the list of credits as a designer in the Convair XP-571 turboprop flying boat prototype.

Ernst Stout's introduction into the country, in July 1933 of supersonic

search using dynamically similar plane models, possibly marks the beginning of higher performance U.S. water-based aircraft. By towing unpowered models and by launching powered flying models—some of which are radio-controlled—he has been able to collect a growing store of design criteria. To this has been added the considerable store of knowledge obtained in model towing tanks such as those at NACA's Langley Laboratory and at Stevens Institute of Technology.

► **Research**—There is space here only for a short roundup of the basic ideas going into the new flying boats, and which now are expected to make it possible to achieve a quality in speed which even the fastest land-based aircraft.

But there are the main points to be considered.

► **Increased ratio of hull length to beam**. This is illustrated by comparison of the hulls of the two XP-571 and the old Navy Convair NC-4 which flew the Atlantic in 1918. Both hulls have 39 ft beams, but the Convair hull is approximately 108 ft long, compared to 45 ft for the earlier Convair hull.

► **Shedding of hull and wing into a** single Mach number aerodynamic form.

► **Use of a newly devised open** up-rear section strip or dam to control effectively all spray without need for any fair, sharp turn, or discontinuity of any kind on the bottom of the hull. The latter design is designed to be restricted down to the hull in flight, leaving a clean, rounded wing and hull cross-section for high speeds.

► **Advantages**—Advantage of the is

crossed hull length-to-beam ratio shows up in better maneuverability and stability plus an increased hull loading up to 100 percent of that formerly carried.

► **Ability to control the spray** without needing hydrodynamic elements under it possible to power the water-based craft with high powered jet engines for the supersonic design now under development.

► **It Means**—And what does that all mean for the future of airplane design? Supersonic flying boat design, and some land-based Navy engineers say that you will be able to do anything with a flying boat that you can do with a landplane, and will have the additional advantage of water-based operation.

The all-purpose high speed plane of the future, then, may be a supersonic seaplane.

► **Shedding Body-Wing**—There isn't too much that one can say about the form of the first full scale flying boat, except that it will be a considerable refinement of the blended hull wing configuration. There is little data available on its specifications or performance, except for a general statement that it is capable of supersonic flight.

► **More Cautious**—But some idea of its next flying boat progress may be gained from a look at Convair's first turboprop flying boat XP-571. The Navy likes the first one so well that they have ordered several more. Recently the prototype made an 8th runway flight and is designed for speeds up to about 400 mph.

It is not now known, that is fast for flying boats. Only one water-based plane in the world has been faster.

► **Fastest Supersonic**—It was a new Italian Mach Conical formplane which was already a flying seaplane.



MODELS of flying boat were launched by hand or ...



CAREFUL as part of hull design tests. Finally, Skate 1 model was ...



TOWED from barge, towing line power boat, with engineers photographing the model of the plane as it cut through water.

A double Fiat engine rated at 3000 hp, whorled a pair of counter-rotating propellers to drive this little speedster to a world speed record of 448.651 mph 16 years ago today, Oct. 25, 1934. Francesco Agnelli, the pilot, couldn't find runway long enough for the plane's takeoff so it was designed with long thin skids which used quite a bit of the increased

length-to-beam ratio principle. It was nearly five years later when a Convair Monocraft 107H plane regained world speed leadership for landplanes. Actually, the turboprop flying boat designed speed is already considerably higher than those of most piston-engine land-based transport flying today.

This is interesting in the light of

the fact that immediately after World War II most airline operators abandoned the flying boat as a transport, except for very special purposes. They took the story back to the beginning. The reason was because the operation all alone, like everybody else, that "supersonic is inherently slower than subsonic."

IATA Meeting

San Francisco convention is organization's first in the United States.

San Francisco—Delegates to the first International Air Transport Association convention held in the United States heard Dr. Albert F. Peters, IATA's president, on problems about the future possibilities of commercial air transportation as impossible.

"Since people think they can already visualize the volume to which air trans-

port can grow," the retiring president of IATA said before he turned the opening meeting here over to the new president, Warren Lee Plessey, TWA board chairman. "In my opinion this is responsible. Technical developments and greater freedom will make air transport to be in part of the question to visualize the volume to which we will come."

Plessey forecast that transportation of passengers will play an increasing part in looking up high density traffic on the world's trunk routes. He said "jumbo" airlines must be paid to develop air freight, suggesting lower rates that would be necessary to build the business. He called the past twelve

months the worst in international airline history.

► **Numbered Problems**—Plessey is his opening address struck out at unorchestrated chaotic volume is representing one of the most urgent problems facing IATA members. He contended that the world's current stand alone and to reduce the price of their services while improving safety, reliability and appeal. But he contended that price reductions must be preceded by a reduction in costs "because an industry in debt is one most unlikely move to a basis of commercial self-support."

A promising outline for solving such is to eliminate or minimize seasonal or

fractional ownership, be emphasized, and he passed the word down to date by IATA in international seasonal and pre seasonal fares.

"But our efforts have been offset by the (usual) of certain so-called charter operators, who have on part of the burden of providing services in the off-season, and who ask us to share the losses. And the operators' trade association has the obligation to protect its members' seasonal businesses and increase the cost of an interrupt in the general public." He urged IATA members to study the problem and to work in "as far as possible to educate the public and their governments as to its danger."

Hilled Deputes—Agreement of Stephen Thorne, deputy secretary general of civil aviation in France, an advisory director of IATA, was announced by the association's executive committee. Thorne will assist Sir Wilfrid P. Hildes in the monitoring and documentation of the association's work for him during the frequent absence. He will head office in Montreal which will work with the association's main office in London.

He will have general responsibility for the work of the IATA traffic conference. To make available compliance with IATA traffic conference resolutions governing international fares, rates and conditions of carriage, as IATA's representative has been established, the committee said. The self-policing nature will permit association members to settle their disputes mutually within the organization, rather than have them appear before the government, the committee pointed out.

Manual Homes—The committee proposed an amendment to the articles of association which would drop the general rule that its headquarters must be located wherever the International Civil Aviation Organization has its office. This would permit IATA office to remain in Montreal in the event ICAO moves to the United Nations Building in New York.

IATA has been invited by the Universal Postal Union to meet jointly with it next January, the committee added, to make an understanding on future air mail rates. Meanwhile, IATA has accepted ICAO proposals for reduced rates for printed matter and newspapers, and agreed for the time being not to reduce foreign air mail cooperation.

Also announced was the formation of a new standing committee charged with the medical and hygienic aspects of matters affecting the safety and efficiency of air travel and passengers.

No Change on Liability—The Warsaw Convention, which governs liability, which holds a "watching hand" on the International agreement drafted in Warsaw in 1929 to set up rules governing air-

craft liability to passengers and cargo, recommended that IATA continue to oppose revision of the Warsaw agreement, now under consideration by ICAO.

The IATA committee pointed out that the many most common device losses have substantially eased the real risk of certain liability in many cases. Any change now in the Warsaw convention would be premature and undesirable, the report stated.

The financial committee predicted that Holy Year traffic would reach the 1959 total of 100 million tourist passengers, with the 1960 IATA clearing house for settlement to \$200 million. The 1949 total was \$166 million, as compared with \$124 million in 1948 and \$12 million in 1947.

Basic operating procedures for voluntary passenger insurance are now being worked out, the committee said. While differences in airline and insurance company procedures presently make a single worldwide standard policy impossible, it believed that it was possible to set up such policies for all airlines operating within certain regions.

Miles and Passenger-Fares miles flown by the world's scheduled airlines in 1949 amounted to 10.5 billion, 10 percent above 1948. The scheduled airlines carried 27 million people, he said, as 3 million more than in 1948, a total of 11 billion passenger miles. This figure was 15 percent above that for 1948.

Even greater increases of income have reported for air cargo, which had doubled since 1948 to \$70 million tons, 20 percent above the previous year.

Meanwhile, he said, safety records had continued to improve.

Trans-Atlantic Traffic—Across the Atlantic, Hilled said, payload between America and Europe had peaked in both directions. The 50,000 air passengers who traveled over the North Atlantic to Europe from October, 1949, to March, 1950, were 51 percent more than the 33,000 who made the return trip. During the second quarter of this year, the total was 15 percent above the same period last year.

In the other direction, where demand has increased from nearly 60 percent in terms of freight, a decrease was reflected at first in passenger traffic from Europe to America, but by the end of 1949 "the effect worked itself out," and this year's first quarter showed a 20 percent rise above the same period of 1949.

After demobilization, air cargo from North America showed an increase of 40 to 50 percent. Shipments from America

to Europe, on the other hand, dropped 18 percent below the corresponding period a year earlier. But they recovered, and in the second quarter of this year were 10 percent above the spring of 1949.

KLM Gets Rights In French Africa

(McGraw-Hill World News)

Amsterdam—KLM Royal Dutch Airlines has received commercial rights from the French government to land at Brazzaville in French Equatorial Africa. KLM will now be able to take passengers to Brazzaville in contrast with the previous arrangement under which it was allowed only to land and refuel at Leopoldville, Belgian Congo. KLM will start using Brazzaville on the Johannesburg-Amsterdam flight from Nov. 15, and passengers from Leopoldville will disembark there and cross the Congo River by ferry—a 10-mile journey.

Heavy Cessna Backlog Poses Many Problems

First Cessna L-19 Hudson plane is expected to be delivered to the Army Field Force late this month, the initial plane in an order for about 900 amounting to approximately \$4 million.

The L-19 (Model 301) jet engine was the competition staged by the AFF and the Air Force, and under the plan contribute the most prominent part of Cessna Aircraft Co.'s military backlog.

This order and expanding subcontract from jet aircraft manufacturers brought Cessna's total backlog at the end of its fiscal year, Sept. 30, to more than \$15 million.

Cessna's two principal subcontractors are with Boeing, for single assemblies for the B-47 Stratofortress, and with Lockheed, for components and air package sections of the B-49 and B-70.

In response to the situation at the close of the fiscal year, Frank A. Boettger, president and treasurer, and "With backlog increasing, our model for production has been altered." Cessna needs skilled workers, and it had had to shuffle its manufacturing order.

Some work done at Wichita was moved to Hutchinson to make use for the B-47 sub-contract work. "As plant work became available at Wichita is handled for Lockheed who demands its military work. Cessna has retained a good location in the market where it is best known, personal as well."

Present annual plane production rate is 130 per month, equivalent to annual sales of more than \$10 million.

PRODUCTION

Government Backs Defense Loans

Program empowering federal agencies to guarantee advances is similar to World War II's V-loan plans.

If your security company gets a defense contract, the government will see that you also get the money you need to fill it. The last time the federal government used your normal line of credit would be permitted to slow up the rearmament program.

The Defense Production Act gave the President power to act up a loan program with the aid of the loan guaranty then government agencies have worked favorably on the cashless for getting out the risk. Some loans have already been made.

Fact Is Factual—The program is going to some financing it resembles the one used at the last war. There, aircraft, engine parts, and accessories were produced and sold at a price of government cost in factures. They received about \$2 billion through the Federal Reserve Board's V-loan program, another several hundred million from the Reconstruction Finance Corp. and \$750 million from the Defense Plant Corp., which built, repaired and leased factories to them.

Federal Reserve has already reached with the V-loan agency. Under it, one of the 12 District Banks is empowered to guarantee all or part of defense loans made by local lenders. The request for a guarantee must be approved by one of the armed services, General Services Administration or the

Department of Commerce, Agriculture or Defense.

Maximum rate of interest a contractor will have to pay is 5 percent, and the local lender has to split the interest with the District Bank. If the payments guaranteed a 70 percent at less, Federal Reserve's guaranteed for only one month of the interest, if the local bank wants Federal Reserve to cover more, the provision for has to pay (in fact, when the guaranteed rate of the loan exceeds 5 percent).

Generally, V-loans will go to subcontractors who can't get advance payments from the armed services—though some prime contractors will get them. The V-loan agency is expected to be used almost entirely for short-term working capital.

No Cash Needed—The only limit to the volume of loan guarantees is the amount the military is going to spend. In effect, a contract with the government is a contractor's collateral. And as long as such collateral is available, the loans will be made. They don't take any cash to start with.

The contractor gets cash directly involved in getting the V-loan. All he has to do is ask for credit at his regular bank. If the bank doesn't think his work or his credit standing warrant a loan of that size, it calls for a guarantee from the Federal Reserve District office.

From there, it's forwarded to Washington.

While waiting for an okay, the District Bank checks the borrower's credit. Meanwhile, in Washington, the Fed and Reserve Board speeds the request over to a certifying officer at the Pentagon, getting approval in a matter of days in many cases. The District Bank then has the money to cash the loan back by phone, and the cash is made available on the spot.

Loan Agencies—Just what if the contractor has to buy equipment or needs a lot of working capital? In World War II, he would have been sent right to the RFC for a direct loan. That time, however, the government agencies would rather see a contractor get into one of the other kinds of aid provided in the Defense Act.

Why? Because the government agencies know that once a contractor has RFC money he can do, they have one control over his operations. They don't expect any change, but they would like to feel that they can just pressure on it if necessary. So they would rather see a contractor first apply for one of the other kinds of aid provided in the Defense Act.

But if the contractor can't get the contractor to get advance payments from the government office his doing with. The money could be filed out in amounts equal to the contractor's performance. Under it, the help only a relatively low price on contracts.

For the subcontractor, though, there is the government's power to purchase goods and services directly from the contractor. So the General Services Administration could help him, say, a tube producer by contracting to buy any part of his output for years to come, then advancing cash against delivery.

For the contractor, the V-loan is less-existing units avoid by the government or new equipment bought directly from the manufacturers. Finally, a certificate of necessity permitting continued construction of new facilities could be the answer.

The kind of help a contractor got would depend on how much he wanted, the purpose and the amount of cash available, the advance payments or the purchase of output and equipment. Only when such funds are fully committed will the contractor be sent to RFC with the certificate that will get him some of the \$2 billion Congress has authorized.

Ohio Contracts

Five Ohio companies have gotten close to a million dollars in USAF business. They are: a contractor in Cincinnati's Ingersoll Rand which received a \$400-



HIDDEN ASSET

Almost invisible under its wartime record of production at the Springfield, Conn., plant being activated by Pratt & Whitney Aircraft division of United Aircraft Corp. is the most modern military aircraft for jet engines. The \$60,000 jet is bulky, built during the war in one of three sub-factories to build monthly production at 260 in East Hartford, but later also was the end of World War II.

977 contract for supplies for the Air Materiel Command. Next was Price Products Division of Bag-Watco, which received two orders totaling \$271,885 for fuel booster pumps ordered by AMC.

Jack & Heintz, of Cleveland, received four contracts amounting to \$394,910 for wing flap assemblies (\$54,748), spread governors \$76,350, aircraft starters (\$58,310) and assemblies (\$57,802).

Other successful contractors were the Calkins Co., Elwyn bicycle manufacturers which has secured a defense solicitation. The contract for a \$31,006 order for aerial delivery aircraft assemblies. Air Flow Corporation Co., Alton, received a \$79,900 contract

Technical Reps in Far East

An activities during the Korean campaign provided a valuable training ground for many pioneer phases and phases of equipment. As a result, the results, and in order to secure production of a number of aircraft components, representatives were assigned to the theater. At one time last month, they were the list:

- Allison Co., CMC, Leonard Division, William F. Kistner.
- Bendix-Stromberg Corp., Alton, California.
- Boeing Airplane Co., Oak Harbor.
- Douglas Aircraft Co., Michael C. Hershman, Ben D. Schell.
- Fairchild Camera & Instrument Co., Easy T. Hunsaker, Robert A. Trull.
- Fairchild Paper & Airplane Co., Vincent T. Gage, S. T. Hahn.
- Hamilton Standard, Inc., Donald A. Smith, Joseph C. Dink, William R. Fawcett.
- Lockheed Aircraft Corp., Glenn C. Feltner, Donald D. Goe, Noble G. Hatten, Ralph T. Stuart.
- North American Aviation, William H. Wagner.
- Pratt & Whitney, Paulsen, Howard L. Smith.
- Wright Aeronautical Corp., Jan Ratz.

Boeing Opens Mechanics School

With the war, have recently available in the Seattle, Wash., area, available to assist the growing need of the Boeing Airplane Co. for skilled and semi-skilled mechanics, Boeing has opened a training school to equip students to perform basic mechanic jobs. Classes will be filled by employees who have completed assembly the company's own pilot indoctrination training course. Forty will be trained at a time, in periods two weeks long, five days a week.

Boeing also has reactivated its apprenticeship training program suspended more than two years ago. Under the renewed program, a five-year program is offered for aircraft and engine mechanics and aircraft mechanic training and a five-year course for tool and die making apprenticeship.

AF Invitations

Back openings are 10 full days after appointment date shown in the following list. Proposals, bid and continuing specifications for items to be purchased will be sent to qualified applicants who state bid invitation number.

One bid set will be available for study before bid opening. Bidders, after bid publication date at each of the areas AMC procurement field offices. That will include time in specification before working in telegraphing for their own bid sets.

Procurement field office locations: Boston Army Base, Boston, Mass.; Portsmouth Aircraft Plant No. 4, Ft. Wadsworth, Tex.; 39 S. LaSalle St., Chicago 5; Wright-Patterson AFB, Dayton, Ohio; West Wing, Wright and Langley AFB, Norfolk, Va.; 115 W. Washington Blvd., Los Angeles, 47; Bond St., N. Y. 4.

Area 1-1 Items: 10 full invitations No. 11-111, issue date 1 Oct., delivery complete within 10 days after date of award.

Area 2-1 Items: 10 full invitations No. 11-112, issue date 1 Oct., delivery complete within 10 days after date of award.

Magador and Comstock 1-10 Items: 10 full invitations No. 11-113, issue date 1 Oct., delivery complete within 10 days after date of award.

Area 3-1 Items: 10 full invitations No. 11-114, issue date 1 Oct., delivery complete within 10 days after date of award.

Area 4-1 Items: 10 full invitations No. 11-115, issue date 1 Oct., delivery complete within 10 days after date of award.

Area 5-1 Items: 10 full invitations No. 11-116, issue date 1 Oct., delivery complete within 10 days after date of award.

Area 6-1 Items: 10 full invitations No. 11-117, issue date 1 Oct., delivery complete within 10 days after date of award.

Area 7-1 Items: 10 full invitations No. 11-118, issue date 1 Oct., delivery complete within 10 days after date of award.

Area 8-1 Items: 10 full invitations No. 11-119, issue date 1 Oct., delivery complete within 10 days after date of award.

Area 9-1 Items: 10 full invitations No. 11-120, issue date 1 Oct., delivery complete within 10 days after date of award.

Area 10-1 Items: 10 full invitations No. 11-121, issue date 1 Oct., delivery complete within 10 days after date of award.

Area 11-1 Items: 10 full invitations No. 11-122, issue date 1 Oct., delivery complete within 10 days after date of award.

Area 12-1 Items: 10 full invitations No. 11-123, issue date 1 Oct., delivery complete within 10 days after date of award.

Area 13-1 Items: 10 full invitations No. 11-124, issue date 1 Oct., delivery complete within 10 days after date of award.

Delaware 1-10 Items: 10 full invitations No. 11-125, issue date 1 Oct., delivery complete within 10 days after date of award.

Area 14-1 Items: 10 full invitations No. 11-126, issue date 1 Oct., delivery complete within 10 days after date of award.

Area 15-1 Items: 10 full invitations No. 11-127, issue date 1 Oct., delivery complete within 10 days after date of award.

Area 16-1 Items: 10 full invitations No. 11-128, issue date 1 Oct., delivery complete within 10 days after date of award.

Area 17-1 Items: 10 full invitations No. 11-129, issue date 1 Oct., delivery complete within 10 days after date of award.

Area 18-1 Items: 10 full invitations No. 11-130, issue date 1 Oct., delivery complete within 10 days after date of award.

Area 19-1 Items: 10 full invitations No. 11-131, issue date 1 Oct., delivery complete within 10 days after date of award.

Area 20-1 Items: 10 full invitations No. 11-132, issue date 1 Oct., delivery complete within 10 days after date of award.

Area 21-1 Items: 10 full invitations No. 11-133, issue date 1 Oct., delivery complete within 10 days after date of award.

Area 22-1 Items: 10 full invitations No. 11-134, issue date 1 Oct., delivery complete within 10 days after date of award.

Area 23-1 Items: 10 full invitations No. 11-135, issue date 1 Oct., delivery complete within 10 days after date of award.

Area 24-1 Items: 10 full invitations No. 11-136, issue date 1 Oct., delivery complete within 10 days after date of award.

Area 25-1 Items: 10 full invitations No. 11-137, issue date 1 Oct., delivery complete within 10 days after date of award.



TWA buys Collins vhf transmitters for its entire Martin 4-0-4 fleet

By its purchase of 40 Martin 4-0-4's, which will begin to go into domestic service next spring, Trans World Airlines follows its traditionally vigorous course of progress.

Also reflecting TWA's policy of providing the most efficient air transportation that modern facilities permit, is the fact that the radio complement of every one of the new 4-0-4's will include a Collins 17L-2 vhf transmitter.

TWA engineers made a careful study of available vhf transmitting equipment. They knew that vhf communication is line-of-sight communication, more-or-less within its range, but not applicable over great distances. They chose the 17L-2 largely because it had the best design weight to power ratio.

The 17L-2, small enough to be housed in a standard 1/4-ATC size case, weighs only 19 pounds. Yet it has a power output of eight watts at 115 volts in a 52 ohm load, thus ensuring that transmitters will be received and acknowledged at the heaviest air terminals.

In addition, it provides frequency control of all 150 channels reserved for aircraft communication in the vhf band. The 17L-2 remote control unit can be located wherever it is most convenient.

All airline operations and communications people should be fully informed about this highly developed transmitter. A descriptive bulletin will be sent you on request.

Collins 17L-2 vhf transmitter

Collins 17L-2 vhf transmitter

Collins 17L-2 vhf transmitter

PRODUCTION BRIEFING

• Goodhue Aircraft Corp., Littlefield Park, Ark., will boost its present working force from 50 to 750 by Jan. 1 in order to handle new orders including plastic and metal aircraft components.

• Conair Aircraft Co., Wichita, has purchased the former Aero Parts plant 41 ac. from the city and will allocate it as additional production. The building contains 100,000 sq. ft. of floor space on a 344-acre tract. Conair currently is building major subassemblies for the Boeing B-47, the Lockheed F-94 and T-35, in addition to having built the L-19 and LC-116 for the services and its general planes.



IN RADIO COMMUNICATIONS, IT'S . . .

COLLINS RADIO COMPANY, Cedar Rapids, Iowa

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27 West Olive Avenue, BURNABY



FINANCIAL

McDonnell Splits Common, 2-for-1

Stock dividend, encouraging broader ownership base, may foreshadow try for listing on national exchange.

The improving fortunes of the McDonnell Aircraft Corp. are reflected in its current action of declaring a 100-percent stock dividend or splitting its common stock two-for-one. Instead of the authorized 560,000 shares already outstanding, there will now be an authorized issue of 720,000 shares.

This stock dividend creates no additional value in itself, as the equity remains the same. In other words, two new shares of the company bear the same value as was possessed by one old share. But stock dividend action is generally prevalent during periods of rising company. Frequently it is a feature of increased cash distributions to stockholders.

► **Heading for the Club-In** the McDonnell outcome, this stock dividend was declared to encourage broader ownership of the company's shares. This may well prove to be a preliminary move in an attempt to obtain listing of the shares on a national stock exchange, probably the New York. Carl McDonnell is the only major shareholder who does not now have such national listing.

Stock dividend action is not entirely foreign to the aircraft industry. General Motors, in June, 1948, declared a 100-percent stock dividend, increasing its outstanding shares from 350,000 to 1,000,000. This not only encouraged broader ownership interest but was followed by increased cash distributions to stockholders.

Borch Aircraft Corp. declared a stock dividend of 50 percent, increasing its outstanding shares from 400,000 to around 600,000 as of Dec. 31, 1948. Constructive results followed this action as well.

Financial credits have emerged that Boeing may soon declare a stock dividend on its shares. The company's earnings are in the thousands and a long continuous run of increasing dividend distributions can be achieved by maintaining the same rate on a large number of shares outstanding.

Douglas with only 600,000 shares of common stock outstanding and selling at the \$50 has frequently been cited as a listed candidate for a stock split up. From all reports, however, the Douglas management has resisted this move. Nevertheless, there is reason to believe that a stock split is in the air there, as

along broader investment interest, would do much to improve the company's market evaluation.

► **Finest War Peak**—McDonnell continues in the upward position of achieving a new peak in earnings surpassing anything recorded during the war years. For the fiscal year ended June 30, 1950, the company showed net earnings of \$2,615,219 as compared with only \$1,751,832 for the previous year—the last year in profit. (Aviation Week Oct. 16). This report of Kern on aircraft production is shared from the June 30, 1950 fiscal results.

During the past year, the company's common stock continued to considerable advance. On June 30, 1949, there were 217,424 shares of common stock outstanding with a book value of \$28.70 per share. Following the stock dividend declaration of \$1 per share on the company's payable May 31, 1950, McDonnell also declared four quarterly payments of 50 cents each, beginning July 1, 1950, or a total of \$2 per share for the 1951 fiscal year.

This had the effect of increasing conversion of the outstanding 3251 per cent shares into common at the rate of two shares of common for one share of preferred for a total of 51,120 shares. The preferred paid \$6 per share annually. The common received a 50-cent dividend per share based on the old preferred.

In view of the dividend policy, it also became advantageous for the owner of common stock to exercise the purchase of stock subject to such rights. Accordingly, a total of 56,265 shares of common stock were acquired at a price of \$50 per share. (The most recent quotation, before the stock split, was in the mid-40s.) The warrants already received, together with an additional 21,738 warrants outstanding, were issued to J. S. McDonnell, founder and president, under an agreement dated July 20, 1939.

The changes in the equity stock, as noted earlier in the complete clarification of the published data together with a selective selection in the number of outstanding warrants, in coming the common shares to 718,229 as of June 30, 1950. As a result, net book value as of that date increased to only \$22.94 per share.

After the 1945 peak in sales of more than \$10 million, McDonnell's sales fell sharply to \$6.5 million in the 1946 period. From that point, deliveries began to mount steadily, reaching \$31.7 million for the 1949 fiscal year and as an all-time high of \$14.7 million for the 1950 current period.

McDonnell's net profit margin as sales, after taxes, for 1950, averaged 5.93 percent. The company, as a feature in its report, however, cites as evidence to the fact that \$377,733 in additional earnings accrued in the 1950 fiscal year, represent a settlement on a completed contract prepared by the management but not yet certified by the government.

► **Backlog Jump**—The effect of the rearmament program is reflected in the sharp jump in backlog figures. As of June 30, 1950, unfilled orders amounted to \$65,685,281. On Sept. 5, 1949, the total backlog was stated at more than \$200 million.

The company's chief customer is the Navy. The single major product has been ordered around an \$18 million share. This was first represented by the F1E Phantom and later by volume orders for the F2H Banshee. The company also has developed the XF-88 jet fighter as a hope of obtaining Air Force orders.

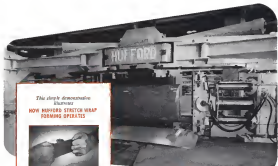
Leaving its main plant facility at Lambert-St. Louis Municipal Airport, McDonnell has been able to avoid any serious property commitments. The modern plant, built by Goetz-Wright during the war, facilitates the overhead banded and is a factor in confining costs to a efficient base.

► **Costs Shrink**—McDonnell should reap increasing benefits from the efficiency inherent in producing one major type aircraft in volume. As the level of activity increases, unit costs should tend to decline.

The company's financial position is the strongest it has been in. At June 30, 1950, its net working capital amounted to almost \$6.5 million. This compared with \$6.5 million a year earlier and as a far cry from only \$29,418 in the same category as of June 30, 1949.

A sustained volume of activity is assured for McDonnell for a number of years to come. At what speed it can accelerate deliveries to meet Navy requirements and at what rate its profit margin remains to be determined. In the past, the company has managed to complete Navy orders ahead of schedule. Subsequent profit margins have been pronounced over the company's previous experience in the leaner development period. The continuation of this pattern would result in the maintenance of high level earnings for McDonnell.

—Sally Altschul



This simple, democratic Hufford

HOW HUFFORD STRETCH-WRAP FORMING OPERATES



Stretch a wide rubber sheet (shown) in a speed which is fully flexible to a wide tape. The rubber is comparable to a metal sheet and the speed never exceeds 100 ft.



New wrap for rubber around the speed mechanical design of stretch film has made it the only conform to the speed design and motion, following the exact contour of the surface. A Hufford press, model also becomes stretch, made readily around the



Unlike rubber, rigidity of metal workpiece is increased when a wide stretch is applied on the die. The final stretch "tuff" the surface, reducing stress of formed part.

Hufford

STRETCH-WRAP FORMING

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Besides the economy of speed, simplicity, and accuracy, the Hufford stretch-wrap forming process provides numerous other advantages to all aircraft and aerospace manufacturers.

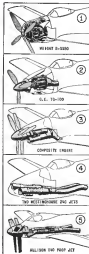
It eliminates or greatly reduces hand operations because stretch-forming is so perfect. • It eliminates hand-forming because every part can be made from stock in the ST condition. • Material is more uniformly stressed over-all since tension is uniform and under careful control. • Field and airplane stretch formers are frequently improved over new material. • Cost can be reduced to extremes of part with greater freedom from problems. • Working and buckling problems are minimized. • Problems difficult to solve by other methods frequently become normal production operations.

Whatever your forming problem—whether it involves extrusions or skin surfaces, investigate HUFFORD—the only machine with the exclusive stretch-wrap forming principle.

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POWERPLANT SELECTION followed the progression: (1) Installation in the AD Skyhawk was a Wright R-3350. (2) Next scheme considered for flight test in AD was a General Electric TG-100 turboprop. Engine was built but the project did not materialize. (3) After this, a composite jet and turboprop arrangement was planned by Douglas, but not built. This scheme was to utilize two Westinghouse 24Cs arranged so that the exhaust would drive a turbine, in turn driving two reduction-gear pumps. (4) Next plan considered was a jet power arrangement utilizing two Westinghouse 24Cs, but scheme was dropped in favor of (5) The final power configuration used in the AD Skyhawk—the Allison T-40 driving axial propellers through a reduction gear.

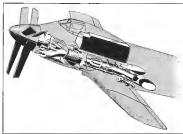


AD SKYSHARK is classic example of how basic features of a power configuration can be modified and adapted efficiently and economically in the changing needs of military service. The new Douglas Navy attack bomber, powered by an Allison T-40 turbo-prop, is a studied evolution of the AD Skyhawk plus powerplant used to produce a plane with a major increase in performance.

to permit it to operate unattended and defend itself against jet fighters. Studies showed that the AD's 2,700 hp would have to be approximately doubled, with an approximate increase in the weight of the Skyhawk (54,865 lb. gross). Result: The turboprop-powered Skyhawk structure was evolved with 1950 hp and a gross weight of approximately 77,000 lb.

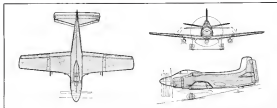
Evolution of the Skyhawk

Turboprop power teams with basic Skyraider design to give Navy attack plane with greater effectiveness.

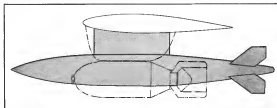


ENGINE INSTALLATION runs under cockpit and is shielded by forward bulkhead, which supports fuel tank. Exhaust ducts to prevent any fuel leak, up into engine area. Jetpipes are put off of pump diff, exhaust aft of tailing edge.

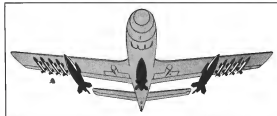
overboard ducts to prevent any fuel leak, up into engine area. Jetpipes are put off of pump diff, exhaust aft of tailing edge.



COMPOSITE THREE-VIEW'S reflecting evolution of AD Skyhawk from AD Skyraider (dashed outline). Wing area and open cruise area for wing and tail thickness are less. Horizontal tail has about same area but has dihedral. Vertical tail area is increased. Height of the top prop plane is about 1 ft. more than on the AD. Cockpit is pushed forward to provide better vision.



NEW BOMB PYLON SHAPE (dashed outline) developed in accord with high speed of AD Skyhawk. Cowl is 16 inches from with the standardized scheme (line with standard 1900 lb. bombs (shown dotted)).



EXTERNAL ARMAMENT SCHEME shows two 2000 lb bombs (outline of new streamlined shape) and eight 5-lb. rockets under outer wings. Alternate bomb locations provide for 1500- or 500-gal. fuel tanks, or even two T-28 rockets.

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Seibel Copter Stresses Simplicity

Newest rotary winger on U.S. market slated to sell for \$11,500. Development expense kept low.

Latest helicopter conceived in the U.S., the Seibel S-4, assumes two clear attributes essential for customer appeal—a rigid simplicity and a low price.

In its conceived form, this rotary winger, built by Seibel Helicopter Co., Inc., Wichita, Kan., essentially is a double-decked flying frame, one level carrying the powerplant, rotor and tail rotor, the other the actual load and pilot.

► **Cost Low**—Development out of the S-4 is reported to be less than \$50,000—greatly below development expense at accorded with other copters previously introduced in the U.S.

Since this expense is reflected in production model selling price, this factor, plus the simple makeup of the S-4, has

worked to make copter's price the lowest in the U.S. helicopter market. Seibel sets the commercial retail price at \$11,500.

First production models are nearing completion. Because of material shortages, future production will depend on the military.

► **Four-Man Team**—Many of the features incorporated in the S-4 were proved in the S-1, a 65-hp experimental copter designed by Charles M. Seibel (now company president) and built in 1947.

Development time for the S-4 was less than two years. Seibel performed the engineering work to complete the design and obtain CAA certification. Richard D. Leisner and George Lab-

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See Brussels National Airport and throughout Belgium, the evidence of Esso Aviation Products in Esso Standard (Belgium).



ESSO EXPORT CORPORATION, AVIATION DEPARTMENT, 25 BROAD STREET, NEW YORK 4, N. Y.

ben, engine mechanics, assisted by John Gibbs, manufactured parts and did all shop work under the development program Pilot Gibbs did the flying.

Sabel's previous work on copters was with Bell Aircraft Corp., on development of that company's Model 30 and 47 rotary wing jobs.

► **Level Deceleration**—Design goal of the Sobel 5-4 is 1500 lb. and actual empty weight 930 lb., affording a favorable useful load/payload weight ratio. Cargo or passenger space on the lower deck is at the craft's center of lift, so are two 16-gal. fuel tanks on the engine (upper) deck. This arrangement is intended to produce no effect on balance in rotation from full load to no load.

Lower area is unobstructed and easily accessible. The deck measures 79 in. long, 36 in. wide and 36 in. high. Level of the deck is but one foot above the ground.

► **Powerplant Data**—Engines are a Lycoming Model C-390-D rated at 825 hp. Power is transmitted to the main rotor through a set of standard automotive spiral bevel gears with a 6.66:1 reduction ratio. In addition to provision for normal starting, hand cranking is provided for alternate use.

Complete powerplant installation—engines, accessories, cooling fan, oil separator, clutch and free-shafting unit, cooling, and exhaust ducts—may be removed in a unit by two men in 45 min., Sabel reports.

Position of the powerplant on the top deck gives practically unobstructed access for inspection and maintenance. ► **Blade-Hub**—Cowman-Burley hub and blade attachment is a highlight in the design's simplification. The hub is cast aluminum and the solid laminated wood blade is connected to it by a piece of flexible steel sheet formed in an angular cross-section. This arrangement allows continuously varying pitch of the blades. The redesign was adopted to eliminate the need for expensive pitch change bearings, and Sobel says that the blade-attaching angle is simple to manufacture, low in cost.

It reports that in almost 500 hours of operation, the rotor hub has required no attention other than inspection and lubrication.

Primary cockpit controls are stand and, linked to the main rotor by push-pull and torque tubes. Rotating portion of the linkage passes through center of the rotor drive shaft.

Two tail bearings and two universal joints are used in the full length of the extension shaft to the tail rotor. The cast aluminum tail rotor gear box reduces standard spiral bevel gears.

Tail rotor hub incorporates same type of blade attachment as does the main rotor. Pitch control mechanism is operated by a cable system connecting to the cockpit pedals.



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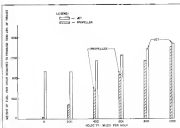
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Fuel weight per hour for 1000 ft. climb vs. thrust for two representative propellers (Edison and Packard), both with the same compressor and burner turbine.

Why We Still Need the Propeller

Detailed analysis of turbo-prop and turbo-jet propulsion factors explains why prop should not be discarded.

By Ivan H. Driggs*

Let's not discard the propeller—yet! At least not here we have examined all the facts. It never pays to jump to conclusions before a close understanding of concepts has been reached.

Rossman and "Buck Rogers" also have not but have no place in the making of final decisions affecting the welfare of a nation. Jet propulsion, including both the turbojet and the rocket, lifts into the dimension category, and undoubtedly has been responsible for the attainment of present high speeds, even above the velocity of sound.

But do we want to go continuously at such speeds? Air men are all well and in their place, but what useful purpose will be served by aircraft designed upon a basis of high speed alone? An airplane must carry something of value and must fly a required distance to be of use. This principle applies to all aircraft whether military, commercial or private, and the type of propulsion, whether propeller or jet, must be chosen in the light of obtaining the greatest usefulness for the time overall.

Propulsion Efficiency—Many comparisons have been made attempting to demonstrate that the jet is a more efficient means of propelling an airplane at high speeds than a propeller.

In such comparisons graphs are

usually used to show the so-called propulsive efficiencies of the two systems. Such graphs which give the ratio of thrust power to shaft power for the propeller along with the ratio of the jet thrust power to the power added to an airstream as it passes through the jet engine fail to provide a true comparison.

Therefore, such data may be misleading to those who are not familiar with the derivations of the two efficiency definitions used above and who do not realize that these two values are, in essence, comparable. In fact, such values should not be drawn on the same graph or even quoted as comparative results. It is hoped these statements will be clear from these derivations:

Let η_p = efficiency of gas not a jet, with no friction losses in nozzle, η_p
 η_j = efficiency of jet, η_j
 η_{pj} = ratio of gas not a jet engine, η_{pj}
 η_{jp} = nozzle efficiency = η_j added η_p with no losses
 η_{TTP} = thrust horsepower of jet.

$$\text{Thrust } T \eta_{TTP} = \frac{2 \eta_{pj} \eta_{jp} (v_2 v_1 - v_1^2)}{2 \eta_{pj} \eta_{jp} (v_2 v_1 - v_1^2)} \quad (1)$$

If the propeller is applied in a gas turbine engine, which is the only way a true comparison can be made, then the thrust power may be expressed as follows:

Let $T \eta_{TTP} =$ thrust horsepower of turbo-jet
 η = ratio of the power absorbed by turbine and gas to the total power available.

least the thrust power would not be as predicted to be zero.
 $\eta_{TTP} =$ shaft power expended in driving propeller
 $\eta_p =$ efficiency of gas not a jet engine, η_p
 $\eta_{jp} =$ efficiency of turbine driving the gas

$$\text{Then } T \eta_{TTP} = \eta_p \eta_{jp} \eta_j = \frac{2 \eta_{pj} \eta_{jp} (v_2 v_1 - v_1^2)}{2 \eta_{pj} \eta_{jp} (v_2 v_1 - v_1^2)} \quad (2)$$

Eq. (2) is concerned with respect to η it will be found that for maximum thrust horsepower

$$\eta = 1 - \frac{v_1^2}{v_2^2} \left(\frac{v_2}{v_1} - \frac{v_1}{v_2} \right)^2 \quad (3)$$

The value of η may be found from Eq. (1) for any given set of conditions and substituted into Eq. (2) for the maximum possible $T \eta_{TTP}$.

It appears that there are three ways the above values of thrust horsepower may be reduced to a comparative basis:

Method 1. Divide both equations (1) and (2) by the power added in propeller through the turbine;

Method 2. Divide both equations by the total power available in the turbine at a given aircraft speed after the gas has passed through the compressor driving turbine, that is, by

$$\frac{2 \eta_{pj} \eta_{jp} (v_2 v_1 - v_1^2)}{2 \eta_{pj} \eta_{jp} (v_2 v_1 - v_1^2)}$$

Method 3. Reduce both equations to values for thrust by multiplying by

$$\frac{2 \eta_{pj} \eta_{jp} (v_2 v_1 - v_1^2)}{2 \eta_{pj} \eta_{jp} (v_2 v_1 - v_1^2)}$$

and then compare the weight of fuel per hour required per 1000 ft. of climb. This comparison squares the use of actual engine characteristics as the basis in algebraic solution.

If the method suggested in (1) above is employed, we have a definition of efficiency for the jet η_j .

$$\eta_j = \frac{2 \eta_{pj} \eta_{jp} (v_2 v_1 - v_1^2)}{2 \eta_{pj} \eta_{jp} (v_2 v_1 - v_1^2)} \quad (4)$$

$$\eta_{jp} = \frac{2 \eta_{pj} \eta_{jp} (v_2 v_1 - v_1^2)}{2 \eta_{pj} \eta_{jp} (v_2 v_1 - v_1^2)} \quad (5)$$

If η_{jp} equals unity (ideal condition) and if a certain η_j , Eq. (4) is an independent variable.

Additional algebraic manipulation as shown by Eq. (4) gives the value of the jet efficiency as unity as η_j approaches η_j (infinity). That is, the jet efficiency becomes 100% when the two velocities are equal and when the thrust power is zero. This does not appear to be a satisfactory definition since, normally, one would expect a maximum value of thrust power when the efficiency is 100%. At

least the thrust power would not be as predicted to be zero.

However, this is the definition of jet propulsive efficiency that is usually employed when comparing the jet and propeller, and has caused much confusion in the thinking regarding the usefulness of the jet. The statement has often been made to the writer that "As you go faster and faster the efficiency of a jet approaches 100% and such a propeller cannot do that." That is true, of course, but one is not interested in propulsive efficiency in the last analysis but only in thrust or thrust power at given jet exit flow. These values should be the ones used.



The new Martin 4-4-4 now joins the long list of aircraft equipped with Edison Fire Detection. Glenn L. Martin's engineers had all the facts before them when they chose Edison. They were well aware of the Edison System's long record of satisfactory service experience by practically every major airline in the United States. They knew, too, of the Edison System's well-earned reputation for dependability, false-alarm-proof design, its low maintenance cost and high efficiency in cutting down lost through delay.

This rightly fitted thermocouple-type system was developed in the Edison Control Research Laboratories. This type of research and development is still going on, striving for new and better products for the aviation industry. Watch Edison!



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Considering the algorithm in (2) above, it appears that a more logical definition of efficiency can be obtained than that derived for Eq. (4). It is this one:

$$\eta_j = \frac{2 \eta_{pj} \eta_{jp} (v_2 v_1 - v_1^2)}{2 \eta_{pj} \eta_{jp} (v_2 v_1 - v_1^2)} \quad (6)$$

$$\eta_{jp} = 2 \left[\frac{v_2}{v_1} - \frac{v_1}{v_2} \right] \quad (7)$$

Again setting η_{jp} equal to 1 and letting v_2 and v_1 approach each other the thrust power approaches zero at the same rate as the efficiency approaches the same value. That is, when the thrust power is zero the efficiency is zero. This makes much more sense and

merely is not as complex as the previous definition. Measurement of Eq. (5a) shows that $\eta_{p,p}$ is a maximum when

$$\frac{\eta_p}{\eta_{p,p}} = \frac{\eta_p}{\eta_{p,p}} \quad (6)$$

$$\text{and for } \eta_{p,p} = \frac{(\eta_p)^2}{\eta_{p,p}} \quad (6a)$$

If η_p equals 100% then the maximum thrust power from a jet is but 50% of the total power available at the hot gas stream and this maximum occurs when the jet velocity is twice the flight speed.

There are very interesting conclusions and ones which are much more realistic and understandable. Further

more, the conception of efficiency given a more direct correspondence with the efficiency of a turbo-prop unit. In this latter case

$$\eta_{p,p} = \eta_p \eta_{p,p} + 1 - \eta_p \quad (7)$$

$$\frac{\eta_p}{\eta_{p,p}} = \eta_p \eta_{p,p} + 1 - \eta_p \quad (7a)$$

Inserting Eq. (3) for η_p

$$\eta_{p,p} = \eta_p \eta_{p,p} + 1 - \eta_p \quad (7b)$$

If a suitable value for $(\eta_p)^2$ of 9% is inserted in equation (7a), and if the product $\eta_p \eta_{p,p}$ equals .475, the last

term in equation (7a) equals zero, consequently the value of $\eta_{p,p}$ will be .475. Comparing this value with Eq. (6a), it will be noted that the efficiency of the jet will be identically the same, i.e., $(\eta_p)^2/2 = .475$. If, however, the efficiency product is greater than .475, the last term of Eq. (7a) becomes negative, and reduces the efficiency, not below .475, but below the value of the product $\eta_p \eta_{p,p}$.

Thus, to use this product as a means of comparing the efficiency of the two systems of propulsion is improper and leads to overstatement of the efficiency of the turbo-prop. However, for a first approximation, $(\eta_p)^2/2 = .475$. If, however, this product may be employed for comparison with Eq. (5a) or (6a), indicating that an unconservative approximation has been made. Eq. (4a) should not be used since the basis of derivation of this equation is not the same as for Eq. (5) or (6). However, the present writer has some comparisons made between the product $\eta_p \eta_{p,p}$ and equation (4a), with considerable confidence. The definition of propulsive efficiency given by Eq. (5) and (7) have been discussed previously in the pages of *Aeronautics* by the present writer, consequently this line of reasoning will be passed on further in this paper.

Third Messure—The third means of comparing the jet and propeller will be the subject for discussion in the balance of this article. If the equation for thrust power derived above are so used to become equations of thrust, then the weight of fuel per hour to produce 1000 lb thrust can be found from the characteristics of any given engine used either as a turbojet or as a turbo-prop. By using the best obtainable efficiency of a given engine, all internal losses that might affect a result are mentioned and therefore a true comparison of two propulsion means can be obtained.

The accompanying chart shows the weight of fuel required per hour to produce 1000 lb thrust for two representative power plants, one a turbojet unit and the other a turbo-prop, both with the same compression and basic turbine. The comparison is given for a series of flight speeds from zero to well above the velocity of sound (750 mph).

Even up to 1000 mph, the propeller will give more thrust per pound of fuel than the jet. At low and moderately high speeds the difference in the fuel required to produce 1000 lb thrust is very great—at 200 mph the jet will consume 34 times as much fuel (600 mph) as the propeller, and at 600 mph, nearly half as much more.

Although the propeller, gear and added turbine will increase the weight of the engine unit, it does not take more hours of flight to make up for this added weight of the power

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plant and then the smaller fuel reservoirs of the poppet-valve engine make for an instant reduction in engine size. In some cases the time of flight may not be long enough that the saving in fuel compensates for the added engine cry, but even then it may be better for some operations to accept a slightly heavier airplane.

Operation from terrain is a case in point. The quantity of fuel that can be stored aboard is limited and refueling requires the carrier to return to base or to a base. A large consumption of fuel reduces either the number of airplanes that can be operated or the number of strikes possible before returning for refueling. Also, the crash-landing thrust at takeoff speeds requires many powerful auxiliary devices for launching. The poppet-valve airplane on the other hand, will take off under its own power with no assistance.

Also, the supply and transportation of fuel to basic area becomes a very critical and possible decisive element in the possession of a base. Can we afford to waste fuel and contribute our own part and supply services by even a small increase in requirements above those absolutely necessary? The fuel supplies in this country are not limitless. Will a strike emergency stand up in face of war under extremely limited or non-existent supply of fuel for necessary military uses?

The fundamental facts as outlined in this article, then, indicate that we must not discard the poppet at all.

Temco Trainer Data

Prototype in American War's article on three competing Air Force trainers (Aug. 23, 1950) was supplied recently by the Temco Engineering & Manufacturing Co., Inc., which has entered the Temco TT-15 business.

Temco says that it is a inherently-weight aircraft that in its entire design line the company designation of TT-15 has been fitted with 165-hp. Franklin engines to step up the Buckeye's performance.

Best change, of course, is a gross weight increase from the earlier figure of 1920 lb. (in the Combatmaster and TE-1A) to the latter of 1975 lb. in the TT-15.

Takeoff ground run is increased from 425 ft. to 610 ft., but the required distance to clear a 50-ft. obstacle has been reduced to 925 ft.

Sea level rate of climb is now 1000 fpm., combining with the TT-15's rate of 925 fpm. At 3000-ft. altitude rate of climb is now 800 fpm., at 5000 ft. it is 665 fpm.

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Details of Sapphire Jet Revealed

New high-power British engine has completed 150-hour service type test at 7200 lb. thrust.

(McGraw-Hill World News)

London—Partial details were released recently of the Armstrong Siddeley Sapphire, claimed by its builders to be the world's most powerful jet engine in its class. Its rated static thrust at sea level is 7200 lb.

With a specific fuel consumption of 9.907 lb./hr./lb. of thrust, the Sapphire is claimed to go a long way toward reducing the problem of high fuel consumption which has so far greatly limited the range and endurance of all jet aircraft.

Armstrong Siddeley's Sapphire has an axial-flow compressor and an axial combustion chamber. This last is a marked change from the multiple array of combustion chambers that has been the usual practice on earlier designs of jet engines.

Sapphire dimensions are as follows: Diameter over hot engine, 35.25 in.; diameter over housing, mounting, 37.5 in.; length overall from front of nose fairing to exhaust cone rear flange, 135.85 in.; jet pipe diameter over hot ahead, 24.5 in.; frontal area, 6.8 sq. ft. Thrust per square foot of frontal area is 1,050 lb., and the net dry weight is 2500 lb.

Further experiment on the thrust of the Sapphire may be expected. While all other performance figures for this new engine are still classified, it is disclosed that the Sapphire has successfully completed the 150-hr. service type test at the 7200-lb. thrust figure. This is 1000 lb. greater than the officially disclosed thrust of the Rolls-Royce and how Avon jet engine or of Rolls-Royce's centrifugal compressor

Turbojet engine, both of which are rated at 6200 lb. (Hunt & Whitney's J-49, formerly selected to the Tex, is rated at 6150 lb.)

One English Four—For some reason at the Sapphire's potential performance in future jet fighters, bombers and civil transports, the builders comment that "one Sapphire has the same power as the four piston engines in a B-29 Superfortress or a Stratocruiser."

This statement was checked with Armstrong Siddeley Motors Ltd., who stated that they based it on the 2500 hp of the Wright Cyclone engine—which have been replaced by PR-9 Whirlwinds in the Stratocruiser.

The Sapphire has been extensively test-flown in the two outboard nacelles of a Lancaster since last January, and in the Sapphire Motor which flew for the first time Aug. 14. The latter craft was flown at the recent SBAC Fair through display.

Mach Number Chart

A Mach number chart, prepared by the Kellogg Instrument Division of the Square D Co., is available to engineering departments of organizations in aviation and allied fields.

The chart presents indicated air speed, Mach number, altitude, static pressure and differential pressure as interrelated functions. A selection of pressure ratio, differential pressure and stagnation pressure is given for various Mach numbers at sea level.

Chart may be obtained free from Kellogg, 8008 45th Ave., Flushing, N. Y.

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Advanced aerodynamic design is what you expect when a top team of designers like those at the Fairchild Guided Missile Division build a missile. But it is only part of the problem in missile manufacture.

A guided missile is packed from nose to tail with complex electronic guidance circuits that must control its flight accurately—even during the shock of launching and the high G loads at tight turns. The engine frame must run "think out" even under loads well beyond those a human pilot can withstand.

In Fairchild's mission, not only the airplane but the complete missile, including the engine design—that are squeezed into the cramped quarters of

the missile's body—are Fairchild designed and Fairchild manufactured.

To gauge the ruggedness and reliability each missile is subjected to G loads never before asked of electronic equipment, while test apparatus shows how the "bell" steps locked on the target. Flight tests confirm the amazing accuracy and ruggedness.

Here is another example of Fairchild research and development, as work for the Armed Services.

Far ahead in the field, Fairchild Guided Missile is an example of the achievement possible when top flight aerodynamics and rigorous electronic equipment tackle closely interrelated problems in a single, united team.



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AVIONICS

Avionic Aids Pay Off for AIL

All-weather flying aids have spelled success for many an unseen flight—and they have also spelled success for the Advanced Instrument Laboratory in Monaca, N. Y.

Airborne gun control is but a part of what can be attributed to an expansion of ways to improve those safety, tactical and landing aids. And now, with the increased importance of all-weather operations, Avionics is devising its time and function completely to government needs.

► **Composite Status**—AIL formerly began business as a corporation in September, 1945. The initial staff was a merger of people from the Radio Research Lab at Harvard, the Radiation Lab at Massachusetts Institute of Technology and the Columbia University Avionics Laboratories Lab. (This last group was originally attached to the National Defense Research Council and Columbia to handle the job of airborne detection.)

Capital for the venture was supplied by the scheduled aircraft industry, because of its stake in all-weather flying facilities were required by the Navy, because of its need for the particular type of electronic laboratory equipment by Avionics.

Capital stock at the outset is owned by Armstrong, Radio, Inc., which in turn is owned by the scheduled airport. AIL currently employs approximately 400 people, about equally divided between professional and technical levels. The ratio of use engineers to technicians is approximately high one.

► **All-Weather Drive**—In general, the field assignments at Avionics revolve around the task of improving all-weather flying. And this means the research, development and limited production of ground-based aids as well as plane-borne antennas and equipment.

Specifically, AIL has performed an evaluation program of the Omega-Bearing Distance navigation system for the Air Navigation Development Board. These tests, prompted by the U.S. proposed acronym by ICAO of the Department, was conducted at three different locations in the country—last year when results differed greatly.

The program included the acquisition of ground-based airborne equipment, the modification of station and maintenance of test equipment and the evaluation of the flight-test program.

During the Berlin Airlift, AIL made

fuel and installed a number of MTI (Moving Target Indicator) sets for the C-54s under its Transport Field, Berlin, Pennsylvania from the low-velocity alongside service crew, meeting not loading with them and the lift pilots.

► **Schneider Avionics—Outstanding among AIL accomplishments is its development of ground-based, multi-angled antennas for high-speed aircraft.**

For VHF and UHF, test cap antennas with vertical polarization have been developed. In the long-range cross navigation band (2 to 24 mi.), the whole airplane is excited to act as an antenna.

But AIL does not produce antennas—it develops the geometry, tests the equipment and then suggests to the manufacturer how the antenna should be built.

► **Phase Tower Test**—It was as an aid to its antenna development that Avionics designed a large, all-phase tower. The tower is a single pylon, 70 ft high, made of a Fiberglas plastic, helically wound. (No metal parts were used in the pylon—ever bolts were made of Monel.) It can be lowered to a horizontal position by a system of gears and a hand crank in order to make installations at the top of the tower.

Small aircraft models (maximum span 5 ft.) are mounted on the top, the tower is tilted to a vertical attitude, and the models are revolved to simulate flight attitudes. A model of the antenna is attached to the tower vertically by a frequency generator and, as the model rotates, the field strength pattern of the small test antenna mounted in the plane is recorded. Whether the antenna is ultimately to be used for receiving or transmitting, its field strength pattern remains the same.

The tower is used to prevent false results, caused by physical proximity, to radio waves from the ground-based frequency. Lines of sight for transmitters and receiver is straight up, not only solving the ground detection problem in the case, but also eliminating ground waves as a factor for other experiments.

► **Defile Record**—The business done by AIL has increased steadily since the formation of the company. A total of \$1,012,567 in 1946 was topped by a figure of \$2,856,019 in 1948.

With its current backlog of \$6 million, business done by Avionics during 1950 will approximate \$4 million. These figures are a high for AIL.

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The ME-43 Dynafocal has an exceptionally low natural frequency which enables it to isolate as much as 95% of engine vibration. The core type spring element has greater oil resistance... is easier to clean and inspect... and is completely resistant to rust. These and other improvements ensure service life and superior performance.

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FAKES KIT enables easier to install disconnect with only minor grinding of accessories

Disconnect Makes Flights Safer

PanAm to install Stratons units which permit safe operation of engine despite generator failure.

By Scott Swings

A new device which makes it unnecessary to cut off an engine because of generator or alternator failure now will be installed on the seven-engine Stratocruisers by Pan American World Airways.

The disconnect will make it possible to pilots, with the flip of a switch, to hot the failing accessory from the engine drive and continue flying on all four engines, without loss of life.

The new engine accessory disconnect has been developed at PAA's request by the Straton division of Hamilton Engine and Airplane Co., Farmingdale, N. Y.

The carrier recently placed a preliminary production order for the new device after 180 hours of service testing. **As Will Be Equipped**—It plans first to install the electrically controlled equipment on all Atlantic division D-77's to make out test "baps" which may appear after extensive operation. Then the unit, possibly modified, will be put in Stratocruisers operated by the carrier's Pacific Area division. Each of the big planes' two alternators and six generator drives off the engines will be provided with a disconnect.

PanAm intends to equip generators on its new DC-8's with these units, also, when these planes are delivered. In all, 28 Stratocruisers and 48 DC-8's are involved in the modification. With

out speeding any particular make, PanAm says all its planes eventually will be provided with generator and alternator disconnects. Where feasible, other engine accessories may in time be similarly equipped.

PanAm has one eye on present needs and the other on growing experience with the kind of equipment for future jet operations, where loss of an engine (which may operate at 40 percent power) at cruise, instead of about 10 percent in both present (propeller-driven) aircraft will be even more critical.

The new disconnect, Model D-3, was specially designed for use with Westinghouse 115m alternators and General Electric SCMT5C1 generator. But Stratons use the unit, with modifications, can be used with other makes and types of engine accessories.

The principle employed in the D-3 disconnect is not new. A similar device, Model D-2A, developed by Stratons over three years ago, is being widely used by L-409 Constellation operates as an emergency disconnect for other power-shaft drive units made by another firm. Also, new Stratons (also single-engine) come equipped with this disconnect as an integral part of the unit.

The disconnect modification requires replacement of the generator or alternator shaft, a slight change in the generator rear section and minor reworking of the housing. The disconnect is designed to make good use of

the external configuration of the generator and alternator used with the Stratocruiser's Pratt & Whitney R-4360 engines. A drop sill or indentation on the front (mounting) face of these units gives ample room for the shaft drive nut device—ensuring the need for housing or extending the accessory for the forward.

► **Fakes Kit**—All parts required for the modification are supplied in a modestly priced kit (shown in left), consisting essentially of:

• A subunit with alternator nut and provisions for mounting it to the generator housing.

• A solenoid-operated trip mechanism and disconnect plunger.

• A special generator or alternator shaft, designed not to fit the spindle and which engages the engine drive.

• A disconnect nut, secured on the threaded shaft and locked by three rivets in a circular steel plate which mates with the engine nut. The result, the disconnect acts to back the generator shaft out of exposure with the engine. Naturally, the large disconnect nut (see right) is not, threaded to the shaft and locked to it by the second lock plate which mates with the engine, is free to rotate with the shaft.

But when the pilot throws the disconnect switch on the cockpit, the solenoid (upper center) actuates the trip mechanism, dropping the plunger (upper right) so that it engages a shoulder on the rotating nut. This stops the nut, completely preventing further rotation. The front plate, riveted to the nut, still wants to go around with the shaft since it is riveted to the engine—and it does. The moving shaft simply tears the plate loose from the disconnect nut, throwing the rivets which lock the parts together.

By breaking the lock which normally secures the disconnect nut to rotate with it, the shaft permits the nut, which it is threaded, to remain stationary, while the shaft continues to be driven by the engine. The result is that the shaft screws itself loosely out of engagement with the engine drive. Clearance is provided, so that as the shaft backs out of engagement, it does not jam against the generator blade tube.

A critical job in development of the disconnect, according to Stratons engineers, was making the front plate so secure against the disconnect nut so that the plate would not shear loose from high torque loads and vibration under normal loads, and thus unlock.

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WHAT'S DOING

at Pratt & Whitney Aircraft?

When World War II ended, just a little more than five years ago, Pratt & Whitney Aircraft had the immense satisfaction of knowing that its piston engines had powered almost exactly as many allied war planes as all other makers combined.

But we were then just about at the bottom of the list as designers and builders of the new gas turbine engines. In fact, we hadn't built a single turbine. The reason? The military demand for our reciprocating types had been so great, right up to V-J Day, that we were not permitted to do more than a token amount of research and engineering on gas turbines. Yet British, German and American jets were already flying, and several other American manufacturers were hard at work on their own designs.

As the war drew to a close, however, we began intensive design and development work on gas turbines. That was in the summer of 1945. We started with the realization that we had a very tough job on our hands to catch up with the others and to stay in business.

But we did it. Just a few weeks ago, we proudly watched the first flight tests of the most powerful turbo-prop engine that has ever flown, the Pratt & Whitney T-34 Turbo-Wasp. Work on this project had started five long years before, under Navy auspices. Meantime, we had already reached the production stage on the most powerful turbojet engine flying in the United States, the J-48 Turbo-Wasp, and on the smaller J-42, both developed in collaboration with Rolls-Royce. In addition, we have continued to refine and produce, in increasing quantities, the Wasp Major, most powerful reciprocating engine in the aircraft world.

Reaching this height was a real up-hill climb, and we think it points up a moral. It's this: Real progress in this aviation business is not easy or quickly achieved.

The story of these past five years at Pratt & Whitney is one of headaches and heartaches and midnight oil, of millions of man-hours of hard work and of millions of dollars risked to reach the goal. The next five years — or ten — won't be any different. We must continue to devote our skill, our energy, our time, and our money to one task — developing superior aircraft power plants. Only by doing this faithfully can we help maintain the air supremacy without which this country cannot survive.

HOW MANY MAN-HOURS HAVE WE SPENT ON GAS TURBINE DEVELOPMENT?



The answer to this question is a very impressive figure and we'd like to publish it here, but to do so I would need instant information. But we can give you a clue to the answer! The simple job to use gas turbine power was development of the J-42 turbojet. That engine, as you know, was only an adaptation of an existing design. And yet more than a million man-hours was spent in preparing it for production. Development work was even more extensive on the powerful J-48 turbojet and T-34 turbo-prop engines. The time devoted to development on all three, and several other gas turbine projects already has run into many, many millions of man-hours.

WHAT ARE THE RATINGS OF OUR MOST POWERFUL ENGINES?



Wasp Major	J-48 Turbojet	T-34 Turbo-prop
<input type="checkbox"/> 3,000 Hp.?	<input type="checkbox"/> 5,000 lb. Thrust?	<input type="checkbox"/> 5,000 Hp.?
<input type="checkbox"/> 3,500 Hp.?	<input type="checkbox"/> 5,500 lb. Thrust?	<input type="checkbox"/> 5,700 Hp.?
<input type="checkbox"/> 4,000 Hp.?	<input type="checkbox"/> 6,250 lb. Thrust?	<input type="checkbox"/> 6,000 Hp.?

The Wasp Major, which went into production with a rating of 3,000 horsepower, exceeds 4,000 horsepower in its latest version and is the most powerful piston engine in the world. The J-48, most powerful turbojet in its category, has a static thrust of 6,250 pounds, but its power is increased considerably when afterburner and water injection are used. And you can be sure that, as development work continues on the power plant, its basic power rating will go much higher. The T-34, although it is in the early phase of its development cycle, is the most powerful turbo-prop now flying. It has officially passed an 80-hour test at 5,700 horsepower, but Pratt & Whitney Aircraft confidently predicts its power will be increased by a very substantial margin.

WHAT TYPES OF PLANES USE THESE POWERFUL PRATT & WHITNEY ENGINES?

- ☐ Bombers?
- ☐ Fighters?
- ☐ Cargo Transports?
- ☐ Interceptors?



The big Wasp Major piston engines are used in many famous planes, flown by the Navy and the Air Force. In addition to Constellation B-36 and Boeing B-50 bombers, Wasp Majors are used in the Douglas C-124 and Boeing C-97 long-range transports, and the Fairchild C-119 and C-119B — all Air Force planes. In the Navy, it powers the Martin Marliner, carrier's tank plane, and the Martin Marmoner, patrol bomber. The J-48 turbojet already has been chosen to power three of the latest fighters in the world — the Navy Grumman F9F-5 Panther, and the Air Force's North American F-86A and Lockheed F94C. The T-34 turbo-prop, of course, is so new that it has only been test flown. But its performance is so promising that already the Navy and Air Force are considering its use in several types of aircraft.

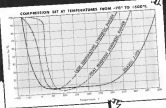


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the test, but would clear easily enough not to damage parts when the internal pressure blasted them out of the test.

■ **Wingless Light**—The pilot's flight engineer is warned to throw as the disconnect switch is a light indicating engine bearing temperature or if that is an indication of radial clearance distribution. It's made certain the aircraft may not be inadvertently operated until the fault has been corrected. The device is designed so that the generator or alternator must be removed from the engine to avoid the shaft for engagement. The disconnect can be used after again after cleaning up and rechecking and replacement of any worn parts.

The device uses accessories from damage in addition to removing the danger of fire when the engine must be kept running in an emergency.

■ **Miding Design**—The mechanical heat, caused by a rotor rubbing against the stator at high speed, in a generator kept in operation after bearings have failed has more than once melted the stator of the generator, so that melted aluminum and copper spread out the coating on each of the stator into the engine accessory system. The high test machine blast from these tests has, on several occasions recently, destroyed the stator stator pressure lines, to fuel transmitter and fuel injector transmitter and spread the oil and fuel in the stator, starting a major fire.

The Civil Aeronautics Administration has expressed interest in the Stator disconnect modification according to FAA engineers. The FAA does not believe installation of disconnects should be mandatory. One engineer told Aviation Week the test was especially suitable for use in craft flying at extreme speeds at altitudes 10,000 feet and above, where the engine, but he could see no reason for its use in domestic airlines with shorter routes and altitudes.

The Eclipse Power division of Bendix Aviation Corp. reportedly is developing a disconnect for its own development. Other large electrical manufacturers are also interested in the Stator design. Some engineering thinking now inclines toward incorporating disconnects, where needed, in the basic design or as an integral part of electrical components.

'Wingless Wonder' Tests Engines

THREE Canada Air Lines' North Star airliners will no longer waste time on the ground running in search of overhead engine. Instead, the two-hour test run will be accomplished on the "Wingless Wonder" made by the test rig of a bent-up International N-7 truck.

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ing, K. W. Carlson and T. S. Skorned, General Electric Co., engineers, discussed the importance of protecting the electrical systems of large multi-engine aircraft.

Their paper was entitled "Distribution System Reliability of 28.5 Volt DC Aircraft Electrical Systems."

The authors stated that a considerable increase in reliability of electrical systems could be obtained by selective minor modifications in line arrangement.

They contended that "a present day aircraft is dependent for best operation on the reliable, continuous service of its electrical system."

The importance of protecting the system as an integrated fault protection design and a well disciplined program of maintenance and continuous cannot be overemphasized.

According to the engineers, the design of a fault protection system should consider three types of fault conditions that involve combinations of voltages of 12 to 28 V, and intermittent arcs.

► **Perceptibility**—They listed these desirable characteristics of a fault protection system.

► **Faults should be cleared quickly** before damage is done to the system components.

► **Protective devices should isolate only the faulted section** and so retain the maximum amount of available feeder and generator capacity.

► **Adequate backup protection** should be included in the system in case one of the protective devices fails.

► **The system should operate without false or nuisance trips.**

► **Conclusions**—After discussing the problems of fault protection in detail, the engineers drew these conclusions:

► "A two-section bus appears to be the best main bus arrangement for ease of fault clearing and general reliability of the system."

► "Symmetrical arrangement of feeders to a forward bus leaves both sections of the main bus results in a better or coordinated system."

► "For the split bus system, battery location at the forward bus gives clearing and coordination of main bus faults."

► **New Circuit Breaker**—Another team of GE engineers, R. S. Ruff and F. J. Reinhold, described a compact and efficient circuit breaker designed to protect aircraft from short circuits.

Planned especially for installation in the fuselage section of multi-engine planes, they stated that the new breaker had ratings of 100 to 600 amperes at 28 V, and 250 amperes at 120 V, that they would operate efficiently from sea level to 10,000 ft, and that they meet the requirements of present-day electrical systems.



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GCA is America's most important environmental aid. It solves most bad weather operational problems: holding stacks, landing and take-off integration, turn-backs, safety checks, proper altitude. GCA ensures traffic at congested airports, "talks down" emergency aircraft in bad weather, provides up to 100% traffic, 300% holding Airport and Airline operations near 100% navigational efficiency. And its share in international relations by making Airbair a 26-hour, all weather operation.

Gilfillan assisted Radiation Laboratories, MPR, to develop the first GCA engineering model. GILFILLAN built the first production type GCA. Since then GILFILLAN has produced the largest number of GCA sets in the shortest time (502 sets in 12 months). Because of advanced design and superior performance, GILFILLAN GCA continues as the standard Radar Landing System at Military and Civil Air Bases the world over!



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For long-term, specifically health-related, studies requiring adjustment of power, sample proportional to study size and study power—from 4 small battery-operated modules are up.



In the present defense program, Budd is engaged in working to help companies to manufacture new products. Budd's Budd Metal Industries Division is now an integral part of American's merged Air Force plants.

Whatever your requirements in the way of search and/or equipment, the facts behind Bendis Radio leadership in this field are important to you—for they tell why Bendis Radio is the logical choice for anything from a tiny range receiver to a complete communications system. Just look at the record. Originally, Research Specialists at Bendis Radio pioneered Radar, since that time they have surpassed even themselves, perfecting G.C.A., V.H.F., Ultra-direction Navigation Systems and many others. Thanks to an engineering

department seemed to sense, the quality and dependability of Bendix Radio equipment has become world-famous; finally, long-range planning and modernization programs have enabled Bendix Radio to build up the largest and finest manufacturing facilities in the entire industry. From every aspect, Bendix Radio's leadership has resulted in progress—new ideas, new techniques, new equipment for you—further proof that it's just common sense to look to the leader for leadership.

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NEW AVIATION PRODUCTS



Eases Tire Handling

"ReGlide," a rubber lubricant designed to promote quicker and safer tire and tube mounting and demounting, reportedly has been adopted by the Latin American division of Pan American World Airways after extensive service testing.

The producer of this lubricant, American Grease Stock Co., says PanAm is the first airline to use ReGlide. Tests at PAA's Miami maintenance base showed that the lubricant does not make tubes sticky or greasy and difficult to remove and handle, according to the maker.

The lubricant is reported to have reduced the danger of damaging tubes and beads when demounting, notes American Grease Stock, adding that others who showed that also ReGlide was used for proper positioning of the inflated tube within the casing, the lubricant also dispensed in such a manner that there was no evidence of tire slippage on the wheel from ground support on landing. Also, there has been no evidence of ReGlide causing rubber deterioration, says the firm. Address: 134 East Erie St., Chicago 11.



Hi-Ratio Gear Boxes

A line of lightweight, high performing gear boxes has been made economically available for the first time by Link Aviation, Inc.

Up to now, these premium devices have been sold exclusively as the latest type training equipment produced by Link for the Air Force. They were built to permit rapid and extremely accurate positioning of accurate reduction and compressor-compressor-in high gear drive logs and in the transfer.

Link originally tried to use gear boxes from outside sources, but none were available to meet the exacting and specific needs of the firm. So it designed its own units. "The Reductive" gear boxes, with gear ratios ranging from 20:1 all the way up to 3125:1.

The units were designed with an eye particularly on keeping friction and backlash to an absolute minimum. Finger pressure alone on the output shaft suffices to turn the input shaft in the 1000:1 ratio gear box—one of the models in the "Hi Reductive" line. During a recent life test, made under a load of 150 in. on with direction reversed each minute, the gear box tested showed no appreciable wear and negligible backlash at the end of a 1000-hr. run, Link engineers report.

The M8T aluminum alloy parts used in these boxes are coated with a special molybdenum compound, furnishing lubrication that penetrates the pores of the metal and lasts for the life of the gear. The dust "protection" had bearings on the shafts are spaced with a special polished oil to insure corrosion-free lubrication at these vital points.

All models measure 3 1/2" x 1 1/2" x 7/32 in. and weigh about 2 1/2 lb. in lots of 500, prices range from \$54 for double reduction gear boxes to \$342 for four shafted reduction models. Address: Hillcrest, Bethlehem, N. Y.

Tube Fittings

Small tube is important tubing around components are "completely eliminated" by the use of "Swagelok" fittings.

So too Crawford Fitting Co. whose engineers have found in tube that the tubing will leak before the fitting leaks. Swageloks are designed to provide a vacuum tight seal and are made in assembly, and eliminate the need for fitted ends on the tubing. They reportedly will serve heavy in this and tubing equally

well and can be used with aluminum alloy, brass, copper, steel, stainless steel and plastic.

Two Swageloks and a threaded chuck made the Swagelok's test, clamp tight around the tubing wall to provide a leak-proof seal. To install, the tubing simply is inserted into the nut which is then tightened 1 1/2 turns. No disengaging torque or twisting motion is transmitted to the tubing since it remains stationary while the nut is being tightened, says the firm.

So occasionally use of these fittings to overcome problems involving pressure, vibration and torque. They are available in brass, Monel, aluminum alloy, steel and stainless steel, and come in various sizes for use with tubing 1/8 in. to outside diameter. Address: 1621 Euclid Ave., Cleveland 15.

Lightplane Refueler

A rugged refueling truck, designed to speed servicing of lightplanes is being marketed by the Hercules Equipment Co., 3893 Olympic Blvd., Los Angeles 24.

The unit, called "Fuel-A-Flow," will pump gasoline into a tank at a rate of 18 to 20 gpm. It has a 220-gal capacity and is equipped with components for bottled and record books.

The Fuel-A-Flow cranks at 10 mph, powered by a 97-hp, opposed four-cylinder Olds CK engine built by D. W. Owen and Sons, Inc., Miami Beach. The engine is mounted in the rear, covered by a fold-back, lift-off housing which mounts the driver's seat, and is located in as far as possible from the fuel dispensing equipment in the front end of the truck.

Rugged, reserve-type transmission is provided to get fully loaded vehicle over bumps, soft or muddy terrain. The truck has two-wheel, 8 x 2 in. heavy-duty automatic hydraulic brakes and is highly maneuverable, the company says.

A diesel unit, the "Label-A-Flow," is also produced by Hercules. This truck supplies engine oil at 5-20 gpm to large aircraft, without need for heating the oil.

Bettors Runway Joints

Closing joints on concrete landing strips is speeded up with new compressors for Model G post closing and grooving machine. This gives more accurate control of cutting bits, whether operating in wet or dry conditions. A tamping frame enables operators to move machine quickly from one job to another. Made by G. H. Trueman Co., 2530 N. Second St., Miami Beach 11.



THE AIRCRAFT: Douglas F3D Navy Skyraider

THE ENGINE: Westinghouse J-34 Jet Engine

THE FUEL SYSTEM: Holley R-46 Turbine Control

HOLLEY
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DETROIT 4



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Our research scientists, engineering staff and production workers offer full cooperation to all aircraft manufacturers, large and small. When you are facing problems that concern aircraft-type Safety Glasses and special glazing instructions for use in long-term tests, Pittsburgh's Pittsburgh Plate Glass Company, 2215-60 Grant Building, Pittsburgh 19, Pa.

As in coated to a temperature of about 50 F., flamed, dehydrated and then introduced into the plant's vault being dried system through a flexible air hose. Uniform air temperature is accomplished by an automatic capacity reduction device built in the mold component. It automatically holds refrigerant output to fixed conditions, regardless of weather. Airspeed varies.

The well equipped air conditioner is housed in a 12-hp. production-built body. Equipment consists of a 7-cylinder radial compressor directly connected to a 150-hp. internal combustion engine. Large cooling coils and an evaporator coil. Auxiliary equipment is V-belt driven and adaptable for belt drive. Flexible pipe connections are

Aircraft Compressor

What is reportedly the first aircraft 5000 psi. high capacity, pneumatic compressor for use in aircraft has been developed by the Convair Co., Missoula, Idaho, among other things, of pneumatic systems for activation of gun charges and bomb bay doors.

The device delivers two cu. ft. of free air at 5000 psi. delivery pressure. It weighs 15 lb. and is available with air or dc electric or hydraulic motor drive. The firm also sells and other accessories for 5000 psi. system also are available. Address, 1141 Metropolitan Life Insurance Bldg., Missoula 1, Idaho.

med between the rubber-mounted engine compressor and the variable component. Water storage tanks are located underneath the track and space is provided for a 200,000-lb. battery to permit year around use. conditioning.

The unit is controlled from a panel located on the outside of the track. Safety precautions include an alarm system which cuts off the system when over three in. loss of engine oil pressure, excessive high engine water temperature, low compressor suction pressure or high compressor discharge pressure. An alarm to the equipment for maintenance is provided by doors on side and back of the track body.

Hot Dimpler Control

A device which automatically can lock into a dimple, used for hot dimpling of aircraft alloy, in long efficient speed production and machine maintenance in aircraft plants.

This modulating hot dimpler control permits die temperature to be fixed rapidly over a wide range, the motor drive. A signal from a temperature-sensitive element is applied to a first stage of magnetic amplification which regulates a main power magnetic amplifier. The magnetic amplifier directly regulates the fluid power applied to the heater unit on the dimpling die.

The controller, for use with 100 to 150 hp. portable or stationary dies, contains no moving parts, eliminates motor moments or vacuum leaks. It is designed to withstand extreme shock, vibration and variations in ambient temperature. The device is made by Industrial Electronics & Transformer Co., 8455 S. Main St., Los Angeles 1, Calif.

Has Pure Iron Rotor

A new type of low inertia rotor motor specially having the highest torque in inertia motor of any size motor now available, has been developed by engineers of the Ford Instrument Co.

This high performance is credited to the use of a pure iron rotor—no place of copper has used. Ford explains the use of pure iron permits construction of a much smaller rotor and elimination of "outgassing" and "single phasing" inherent in some copper bar types.

The new 18-watt motor, designed to comply with military specifications "permits new class of zero maintenance," Ford says. Since diameter of the rotor is only 0.665 in., compared to 1 in. is required in the former model, the rotor is lighter, having a saving of one lb. in overall weight has been achieved, accompanied by a noticeable increase in the torque to inertia ratio.

The motor has linear torque speed

characteristics over the entire operating range and motor speed of 15,000 rpm. is governed either from single or two phase, 115v., 60c. current and can be flange or foot mounted. It weighs 4.3 lb. Address, 5110 Wisconsin Ave., Long Island City, N. Y.

ALSO ON THE MARKET

Tubing is protected by new Tensite plastic caps that snap over beaded ends and provide a positive, dust-proof seal. Available in various colors, caps reportedly can be used over many times without losing sealing properties, will not loosen from ordinary handling, at tubing. Distributed by Hall Industries, Inc., 111 W. Jackson Blvd., Chicago.

Model 299 pressure transducer for measurement of pipe or differential pressure in the range from 0.1 to 9.20 psi. also is offered as an absolute pressure transducer in the range 0.15 psi. Absolute response is less than 0.1 percent of full scale per G up to 10 G. Address, 10000 S. Main St., Station Laboratories, Inc., 9728 Santa Monica Blvd., Beverly Hills, Calif.

Instrument for measuring amplitude of machine vibration is held by hand against the machine being checked. A detachable 6-in. probe, supplied with vent, Model P Vibrator, accurate measurements, indicated on built-in scale, to be taken directly on machine or a reading plate. Made by The Vibroscope Co., 57 E. 79 St., New York 16.

Small dual pumps for track tractors, intended for remote wheel delivery systems, don't require 50 gpm., is suitable for use with gasoline and other liquid hydrocarbons having viscosity up to 120 S. S. Made by Marlow Pumps, Ridgewood, N. J.

Starling cable for fractional horsepower, single-phase motor apparently has wide application where it can be used to advantage over conventional centrifugal motors. Having a balanced armature construction, the unit is not subjected to one wire in one position only. Made by RHM Division, Essex Wire Corp., Longmont, Ind.

Soldering gun spoolholders with, feature biggish growth and new thermal energy need designed to better heat shield-shaped soldering tip and remove metal chutes from final work. Useful in tasks such as work done in higher and greater models have dual heat (100/175 W.) and two-half-inch gap built ahead of soldering tip. Made by Wells Electric Corp., Eureka, Pa.



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PITTSBURGH PLATE GLASS COMPANY

AVIATION WEEK, October 29, 1996

Pictorial Computer works, look first at the other equipment on which it relies.

► **Operation**—Everything is computed electronically—where you are with reference to the single GED station at any, Updon, Utah—and where you want to be.

If you crank in the proper settings on the Course Line Computer, you can fly straight to Updon Airport, simply by following the cross-pointer needle on the instrument panel centered.

Install a Pictorial Computer instead of CLC and you do not have to pre-select your course. On Pictorial you may change course as often as you wish. You don't re-align your new course to destination or crank in new coordinates.

In essence, all the Pictorial Computer does is put a pencil in the hand of a Course Line Computer and stick a map under the electronically guided pencil.

Actually, CLC and the Pictorial use their bearing-distance information in different ways. On CLC you figure two things: range and bearing to destination from the ground station, and the magnetic course you want to fly to that destination. Then you crank those figures into the CLC. Its needle gives bearing to destination from the course selected.

► **The Bag and Toss-On Pictorial**, all that happens is that the computer keeps recording your present range and distance from the station all the time. The station is the center of the chart and the bag is you. Range and bearing from the station is the start of the bag's range and bearing from the center of the chart.

If you have a Pictorial, you do not necessarily need to buy Course Line, too, although you could navigate a little more precisely on a straight, preselected course using the Course Line instrument.

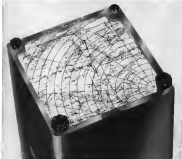
► **Error Mechanisms**—Range of overall error on Pictorial Computer instruments, including error of ground station and onboard VOR and DME, is a maximum of ± 15 degrees azimuth and one-half mile distance.

Close in, where it counts most, the track distance error is greater than azimuth error in terms of miles. So the maximum error from dead center is ± 4 miles.

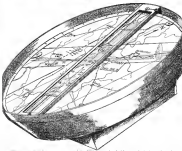
You can count on getting a spot one mile away, says ANDR.

Pictorial Computer works with all three official enroute map scales: world maps charts (1 in = 16 statute miles), sectional aeronautical charts (1 in = 5 mi.) and instrument approach charts (1 in = 4 mi.). On these charts, the coverage sector of the computer takes in about one diameter of 160 miles, 80 miles and 40 miles respectively.

CAA will take in standard enroute maps and merely cut them to fit the Pictorial Computer. Each piece will be cut for use with the particular GED



CONSOLE Pictorial Computer shows position by the moving crosshair and scale on map.



PORTABLE Pictorial Computer weighs little, can be held on pilot's lap when in use.

ground station of an airport or area. But once here, use will not be limited to one way or station in a crowded area.

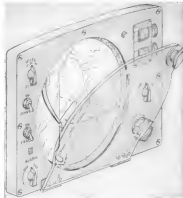
You can hit a Washington airport or objective across the state, the Washington map and GED range on the Baltimore set. The same is true for other areas.

► **Portable-Pictorial**—Here is a rough description of the 5322 Portable Pictorial Computer ANDR aviator has set the general specifications for

he can not set the exact functional design. Makers of pilot testing will decrease final model.

The Portable is a flat box designed for mounting anywhere in the cockpit when not held on the pilot's lap for observation. Weight is probably around 10 lb., certainly under eight. Thickness is 1 1/2 in.

The portable has a round lens, with an overall diameter of 11 1/2 in. Mapview



PANEL Pictorial Computer, being made by Sperry, contains essential elements of CLC.

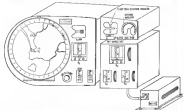


FIGURE 1, Course Line Computer and aircraft wiring which can be separately located.

ing before is also round, with diameter of 15 in. A transparent plastic cover tops it, to protect the sensitive of the course wiring from damage.

ANDR says the gadget will take normal cockpit handling without losing its built-in accuracy of ± 1 degree of azimuth and 3/4 statute miles.

The whole box is simply dinged-up maps to which on the pilot's engine console or control. Only attachment is the need to a pink box for the GED pin-

The position-indicating bag is a small bag. Through the hole in it you can insert a pencil to track on the map, or you can drive the pencil out if there's no need for a pencil track (Permanent tracking on the panel and console models may be done automatically in one of several ways—such as invariable pointing—but ANDR hasn't yet decided which way).

The Portable is like the other models in its general use and operation, only

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simply. They all have a magnetic-lead- ing indicator scale that goes all the way around the 10-in. rotor's circumference. Because at the top scale of this heading indicator, the pointer is easily read by the pilot to within half a degree of actual heading.

► **How to Fly It.** You navigate with the Portable in the same general way as with the more elaborate Panel and Console models.

If you fly from San Francisco to New York, you will probably start out by plotting the San Francisco area chart in the Portable shortly after takeoff. The local OBD station is in the center of your map.

Then the heading scale to the frequency of that OBD station, and switch on the gear, selecting the proper map scale for the map.

The position line immediately moves to your present position and starts tracking your plane. By this time, you are probably several miles east of San Francisco. The map is accurate to the right on your chart as you climb upward on course. If you are an overnavigator, you will perhaps use practice using the Portable because you can compare the sometimes with more certainty of position and course. Then you can before you find the Portable Computer.

You may use your Portable while other pilots use the clear, but it will still track you.

A flag shows exactly up when you fly beyond the range of the OBD station you have tuned the Portable to. That the flag is just like the ILS area pointer flag shows indicator.

The other three are:

- **Flare** shows outside the range of the OBD station the Portable is tuned to (range is limited by DME to optical line-of-sight).
- **Interference** sets into the receiver.
- **Procedural** flag.

If you find the flag shown is up, and the flag has come up toward the side of the cover frame, you know you should switch to another map. You know also that when you left the limits of the outside outlined chart your course was that indicated by the trail of the flag.

If you find the flag shown is up while the flag is still tracking a major course on the map, you are probably behind a mountain and have broken your line-of-sight contact with the ground station you are tuned to.

If the flag shown is up and the flag is behaving erratically, something is wrong.

To switch maps, look at the next map. Turn the Portable to the OBD more indicated, remove the old map, and set up the new. Navigating transcontinental flight using Portable Computer would require about 15 map changes if you wanted to see

your position at all times. If you want to land at Chicago, you may switch from its area chart to a national about 40 miles out and then perhaps to an expanded map at about 50 miles out. ► **Steady at High Speed.** Let plane speeds are where the Portable will come in most handy. Portable is undoubtedly the simplest way to compute at high speed because even contact conditions give the pilot little help, since he moves too fast to keep track of his calculations.

At jet speeds, of course, you have to change a small chart every 15-20 minutes, but each switch takes only 15-20 seconds.

Portable Computer will probably be favored over Console and Console, but only pilot selection will tell actually. Production points of Panel and Console is estimated at \$100 each.

► **CLC Databank.** This is an apparatus where disadvantages of Console Laser- ► **Shifting course** on Console Laser Computer requires you to compute a new set of coordinates as you turn into the instrument. This not only takes time but also allows room for pilot error. Same risk exists every time you switch course on CLC.

► **Flare the scale.** on CLC gives only a left-right deviation, plus a figure on distance, whereas on Portable you have the "lead" of where you are going. Only a map pattern or outline (light) can give you that.

► **Map reading** is still necessary when you use CLC.

On Portable, you fly your course and read your map all at once. Portable's operation—adjust, plot, change, use, and change of error. It is in line with today's trend toward "computerized" instrument groupings (see Aviation Week Oct. 5, p. 47—"Bread- and-Butter One-Stop Map").

► **Panel Model.** This is the more elaborate Panel and Console models of the Portable Computer the same as the Portable. But these have added features.

The Panel has a magnetic compass heading error mounted on top of the moving position-indicator flag. This extra heading indicator is supposed to help correct the pilot as he glances at the map. But at the accuracy of the magnet-controlled arrow is so poor (1-7 degrees), it hardly seems worth the effort.

On the Panel you can change the north-south alignment of the map. This, while you install the map in the mechanical way, with north up you may rotate the chart with a special control. So you may fly your plane "up" the chart whether it be north, south, east or west.

The Panel model is not portable. It is designed for permanent mounting in the cockpit, probably in the instrument panel.

Frontal area is about the same as the Portable (just two inches wider, same height). Depth is not over eight in., so it is suitable for instrument panel mounting. Weight about 20 lb.

The Portable is available alone, or combined with the two Console Laser Computer flight instruments and/or CLC controls.

► **Console Model.** The Console is a miniature. You dial the identity code of the range you want and the chart flashes on the screen (projected from inside). Thus there is no mirror map screen. The same dialing has already been done in the proper range for the map.

Your position shows up on the Console's screen as a gull, or vehicle. Your position is surrounded by radial lines giving you bearings, and concentric circles giving you distance—on a 100 degree scale screen. As the plane moves across the map, the radial and concentric lines, too. The radial scale under the influence of a magnetic compass.

Weight of this equipment is under 25 lb. Screen is less than 14 in. dia. with probably about the same frontal area as the Panel and Console models, but deeper.

Inside the Console is a projector with a 15-foot dia. lens. There are 170 charts of your disposal—weatherable ANDR-1000. This is simple map section that the main reason of an airline.

► **Interference.** Single-Sense questions about Portable's portability come to mind.

► **Will two stations** on the same frequency interfere the Portable's little bug? No, as long as you stay on your map.



NEW SCOREBOARD

This new glass scored departure duty until board being installed on all Pan American World Airways transports is replacing present lumber-tooled blackboards. The new board has interlocking lock letters and numbers for easy visibility.

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- 2. **WEIGHT** lightest of its type, 2.2 ounces
- 3. **PRICE** lowest priced, \$1.95 in type 10-0
- 4. **RELIABILITY** longest service life of type—40,000 hours
- 5. **CONSTRUCTION** hermetic sealed

Plus these other specifications

- 1. **CONDENSER RATING** 1A, 25V, D.C., 1A, 15V, A.C., 400 cycles
- 2. **CONTACT OVERLOAD RATING** 10 A, 15 V, 20 am
- 3. **ATTITUDE RATING** Dry, wet, gas, pressure, shock, barometric, etc.
- 4. **COIL RESISTANCE** 300 and 100 ohms
- 5. **COIL VOLTAGE** 10 V D.C., 150 V A.C.
- 6. **TERMINAL RESISTANCE** 10 ohms
- 7. **WORKING** Variable
- 8. **VARIATIONS** Virtually limitless, in voltage, frequency, number of poles (to 100) and temperature.

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- To solve a difficult penetration problem on a bomb handle, Hoover designed this actuator which is controlled by a direct pressure measuring device through a potentiometer. It includes a 7500 lb. 1/2 turn planetary reduction gear, and the motor shaft can be loaded at simply five times the output that is fitted to 75 degrees.
- This actuator was designed by Hoover to withstand the extreme loads encountered during in-flight adjustment of the horizontal stabilizer of a jet fighter. The actuator weighs only 32.5 pounds, yet will withstand an ultimate load of 15,000 pounds without failure.
- For positive locked coupling, Hoover built into this power unit a unique wedge locking device with no adjustment parts, the load being increased by spring-loaded ratchet. It has operated 1,000 consecutive times with a mean time to failure in excess of 100,000 hours specified value.

HOOVER

ENGINEERING

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there can be no normal interference from other stations on the same frequency.

• Will "skip," or break, reception of broadcast signals easily? Yes, just as it can expect any other electronic receiver. But the skip signal must be stronger than your local signal. Odds are there's no interference. OADR says.

One of the good things about the trend in short circuit movement of the leg leaves a pointed record. So you wouldn't have to be watching it all the time to learn that the ship had arrived.

• Is it reliable? Yes, it should be as reliable as ILS or any other such equipment. This all depends on careful engineering design and construction. Federal Commander is not a radical idea—it is an application of long-known principles and components.

• When Can You Buy?—If all goes well, Federal Commander, Control Line Decoder, and DME will be available about mid 1952. Then the airlines will probably take a year out of the new navigation gear in their planes (at over half price).

Big catch right now is that the airlines haven't decided on specifications for airborne DME. That's because the last CAA is already being delivery on 34 airborne DME's from Federal Telecommunications. Let's fill the bill, for large scale evaluation.

Ground equipment is already going into place. CAA has 450 ground DME units (manufactured on order from Radiovac Electronics Corp.) for installation starting next March.

Airfreight Skirts Inventory Risks

The Skender-Lock, Skut Co., Philadelphia, has worked out a system of airfreighting stock to Atlanta, enabling them to keep full inventory control with a minimum investment in goods. American Airlines does the flying of the freight.

The retailer anywhere in the U. S. buys his inventory agents of stock for only a week's business. On each skirt is a duplicate stock tag.

Every time the retailer makes a sale, he tears off the tag and sends it to the manufacturer in Philadelphia. The manufacturer is to fill another order within a week.

American offers one-day delivery to Skender-Lock's retailing out of the Mississippi, and second day delivery elsewhere in the U. S.

Skender-Lock's President, Carl H. Skender, calls his system "riskless merchandising." It helps overcome the usual retailer fear of stocking seasonal or inventory of goods.

Continental Tests Omni-range System

Continental Air Lines pilots last week started using the first official VOR "Omni-range" controlled airport system. The new VOR high frequency radio range system covers 7150 miles of air ways. Continental will give VOR the test run. The device class many advantages over the old low-frequency, low-power range system. Among the advantages are: little static interference, bearing identification from any point within range of a single station, a definite position "fix" within range of any

two stations (requires plot calculation), visual orientation on instrument, instead of total readout on instrument.

Pilot navigates directly toward or away from the station, by entering a number, which gives kilohertz deviation from the bearing course.

The air mail pilot of the 1920's, had to see the ground to keep to his track. In the 1950's he could fly his airplane at a greater speed along route—about, but well defined, sky trails, marked by the beep-beep of the "radio range" in his earphone. Trouble was, when he needed the range most, in a storm, the static on the low-frequency radio nearly drove him mad. And if he

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Morton, Pa. (Over Philadelphia)

Allison FIRSTS — Proof of Turbine Engine Leadership



T38 Turbo-Prop

FIRST to complete 150-hour qualification test for Turbo-Jet engine with afterburner.



T40 Turbo-Prop

FIRST to complete 150-hour qualification test for Turbo-Jet engine with water/alcohol injection.

FIRST to complete 150-hour qualification test for any Turbo-Jet engine.

FIRST to complete 50-hour flight clearance test for U. S. Turbo-Prop engine.



J33 Turbo-Jet

FIRST to put Turbo-Jet engines in production with either afterburning or water/alcohol injection.

FIRST to purchase its own airplane to proof-test Turbo-Prop engines for commercial transport use.

FIRST to design and release for production a Turbo-Jet fuel control which meets combat operational requirements.

FIRST to get commercial certification of a Turbo-Jet engine.



J35 Turbo-Jet

FIRST to fly a propeller-type engine producing more than two horsepower per pound of weight.

Allison

DIVISION OF



INDIANAPOLIS, INDIANA